



**HEAVYWEIGHT HIGH IMPACT (H.I.)  
SHOCK TEST REPORT**

**July, 2003**

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### **3.0 TEST REQUIREMENTS**

#### **3.1 Shock Specification**

MIL-S-901D (U.S. Navy)

#### **3.2 Shock Grade**

Grade A

#### **3.3 Equipment Class**

Class II

#### **3.4 Mounting Location**

Deck

#### **3.5 Mounting Plane Aboard Ship**

- 1) HF Receive – Back and Base
- 2) System Control – Back and Base
- 3) V/UHF Receive – Back and Base
- 4) CCOP – Back and Base

#### **3.6 Mounting Orientation Aboard Ship**

- 1) HF Receive – Unrestricted
- 2) System Control – Unrestricted
- 3) V/UHF Receive – Unrestricted
- 4) CCOP – Unrestricted

#### **3.7 Number of Articles**

4

### 3.8 Items Tested

#### 1) HF Receive Cabinet Configuration

Manufacturer:

Part No:

Shock Isolators: Base – (6) 901D Part No. 70512-1  
Sway – (2) 901D Part No. 70526-1

| Qty. | Hardware Item                |
|------|------------------------------|
| 1    | AN/SSQ-137 (V) 1 Rack        |
| 1    | Patch Panel HF               |
| 1    | Distribution Unit, RF        |
| 1    | Printer Unit, Laser          |
| 1    | Tuner, HF                    |
| 1    | Workstation, Intel (Jots3)   |
| 1    | Keyboard/Display Unit        |
| 1    | Digital Processing Unit      |
| 1    | Uninterruptible Power Supply |

#### 2) System Control Cabinet Configuration

Manufacturer:

Part No:

Shock Isolators: Base – (6) 901D Part No. 70512-1  
Sway – (2) 901D Part No. 70526-1

| Qty. | Hardware Item                     |
|------|-----------------------------------|
| 1    | AN/SSQ-137 (V) 1 Rack             |
| 1    | Time Unit, Precision              |
| 1    | Distribution Unit, Time Frequency |
| 1    | Filter Unit, HF, V/UHF            |
| 1    | Interface Unit, INT/EXT           |
| 1    | Patch Panel Audio                 |
| 1    | Receiver, R-2411                  |
| 1    | Up Converter                      |
| 1    | Workstation, Intel (Jots4)        |
| 1    | Keyboard, Display Unit            |
| 1    | Server, Unix (CUB)                |
| 1    | Server Unix (LH)                  |
| 1    | Uninterruptible Power Supply      |

### 3) V/UHF Receive Cabinet Configuration

Manufacturer:

Part No:

Shock Isolators: Base – (6) 901D Part No. 70512-1  
Sway – (2) 901D Part No. 70526-1

| Qty. | Hardware Item                |
|------|------------------------------|
| 1    | AN/SSQ-137 (V) 1 Rack        |
| 1    | Patch Panel, V/UHF           |
| 1    | Distribution Unit, RF, V/UHF |
| 1    | Receiver, R-2412             |
| 1    | Receiver, V/UHF              |
| 1    | Workstation, Intel (Jots5)   |
| 1    | Keyboard/Display Unit        |
| 1    | Digital Processing Unit      |
| 1    | Uninterruptible Power Supply |

### 4) CCOP Cabinet Configuration

Manufacturer:

Part No:

Shock Isolators: Base – (6) 901D Part No. 70512-1  
Sway – (2) 901D Part No. 70526-1

| Qty. | Hardware Item                |
|------|------------------------------|
| 1    | AN/SSQ-137 (V) 1 Rack        |
| 1    | Uninterruptible Power Supply |

## 3.9 Overall Equipment Dimensions/Weight/Center of Gravity

### 1) HF Receive Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)  
Weight: 724 pounds  
Vertical CG: 34" from mounting base

### 2) System Control Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)  
Weight: 820 pounds  
Vertical CG: 34" from mounting base

### **3) V/UHF Receive Cabinet**

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)  
Weight: 824 pounds  
Vertical CG: 34" from mounting base

### **4) CCOP Cabinet**

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)  
Weight: 297 pounds (dummy loaded to 800 lbs. for testing)  
Vertical CG: 34" from mounting base

## **3.10 Mode of Operation**

The test units were powered and operating for all detonations. The equipment under test (EUT) was powered with 110 VAC.

## **3.11 Shock Test Acceptance Criteria**

In addition to the acceptance criteria presented in Section 3.1.10 of MIL-S-901D (Navy), the following was also considered as acceptance criteria for meeting the shock test requirements:

1. The test item continues to perform its intended functions following the test series.
2. A momentary malfunction shall be considered acceptable if it is automatically self-correcting.
3. No part of the test item that may be considered a hazard to personnel or Grade A equipment shall break or come adrift.
4. The test item shall not demonstrate a potential for fire hazards. Any evidence of electrical shorts, release of flame, smoke, or sparks shall be cause for rejection unless otherwise approved by the acceptance authority.
5. Minor physical damage to the test item, such as small cracks, minor yielding of structure, out-of-tolerance clearances, and similar damage shall not be cause for shock test disapproval unless such damage causes unacceptable impairment of equipment performance, results in a hazard, or results in a substantially shortened equipment useful life.

### 3.12 FSP-Borne Weight

|                                  |                      |
|----------------------------------|----------------------|
| Test Items                       | 3,168 pounds         |
| Test Fixtures for Class II Items | 3,573 pounds         |
| Other Items on DSF               | 3,386 pounds         |
| DSF                              | 11,000 pounds        |
| Instrumentation and Rack         | 1,200 pounds         |
| Canopy                           | <u>7,000 pounds</u>  |
| <b>Total Weight Borne by FSP</b> | <b>29,327 pounds</b> |

## 4.0 TEST METHOD

### 4.1 Test Facility

The test series was conducted at the facilities of Dynamic Testing (DTI). This test facility is approved for testing in accordance with NAVSEAINST 9491.1B and by NAVSEA Letter 55X11/SH, Serial 87, dated 17 February 1984.

### 4.2 Test Series

The test series was conducted in accordance with MIL-S-901D and 901D Document No. D00231-QTP-001, using the FSP, constructed in accordance with BUSHIPS Drawing No 645-1973904, utilizing standard, 60-pound HBX charges suspended at a depth of 24 feet below the surface of the water. Standoff distances were measured from the center of the charge to the FSP vertical hull plating.

After Shot No. 3, the test items were rotated 90 degrees in order to meet requirements for unlimited orientation.

**Table 1. Test Series**

| <b>Shot/MIL-S-901D Test Number</b> | <b>Standoff Distance</b> | <b>Deck Frequency</b> | <b>Charge Location</b>                      |
|------------------------------------|--------------------------|-----------------------|---|
| 1/2                                | 30 feet                  | 14 ± 2 Hz.            | Along the athwartship centerline of the FSP |
| 2/3                                | 25 feet                  | 14 ± 2 Hz.            | Along the athwartship centerline of the FSP |
| 3/4                                | 20 feet                  | 14 ± 2 Hz.            | Along the athwartship centerline of the FSP |
| 4/4 Rotated                        | 20 feet                  | 14 ± 2 Hz.            | Along the athwartship centerline of the FSP |

The fundamental vertical response of the DSF was determined from the peak response on a fast fourier transform (FFT) generated from acceleration time-history data measured at Sensor A1V (see Photograph No.3). Deck frequency and support locations are summarized in Table 2.

**Table 2. DSF Frequency and Pin Support Locations**

| <b>Shot/Test No.</b> | <b>Sensor No.</b> | <b>Frequency (Hz.)</b> | <b>DSF Pin Support Locations</b> |
|----------------------|-------------------|------------------------|----------------------------------|
| 1/2                  | A1V               | 13.1                   | A5/D5                            |
| 2/3                  | A1V               | 13.1                   | A5/D5                            |
| 3/4                  | A1V               | 13.1                   | A5/D5                            |
| 4/4                  | A1V               | 13.1                   | A5/D5                            |

**4.3 Exceptions to MIL-S-901D**

Instead of the normal sequence of shots specified in MIL-S-901D, the shot sequence listed in Table 3 was conducted in order to meet requirements for unrestricted orientation.

**Table 3. Planned Shot Sequence**

| <b>Shot No.</b> | <b>MIL-S-901D Test No.</b> | <b>Charge Standoff (feet)</b> |
|-----------------|----------------------------|-------------------------------|
| 1               | 2                          | 30                            |
| 2               | 3                          | 25                            |
| 3               | 4                          | 20                            |
| 4               | 4 (rotated)                | 20                            |

**5.0 TEST INSTALLATION/CONFIGURATION**

**5.1 Equipment Installation**

Photograph No. 1 shows the installation for Shots 1 through 3.

Photograph No. 2 shows the installation for Shot 4.



**Photograph No. 1. Overhead View of Initial Installation, Shots 1-3**



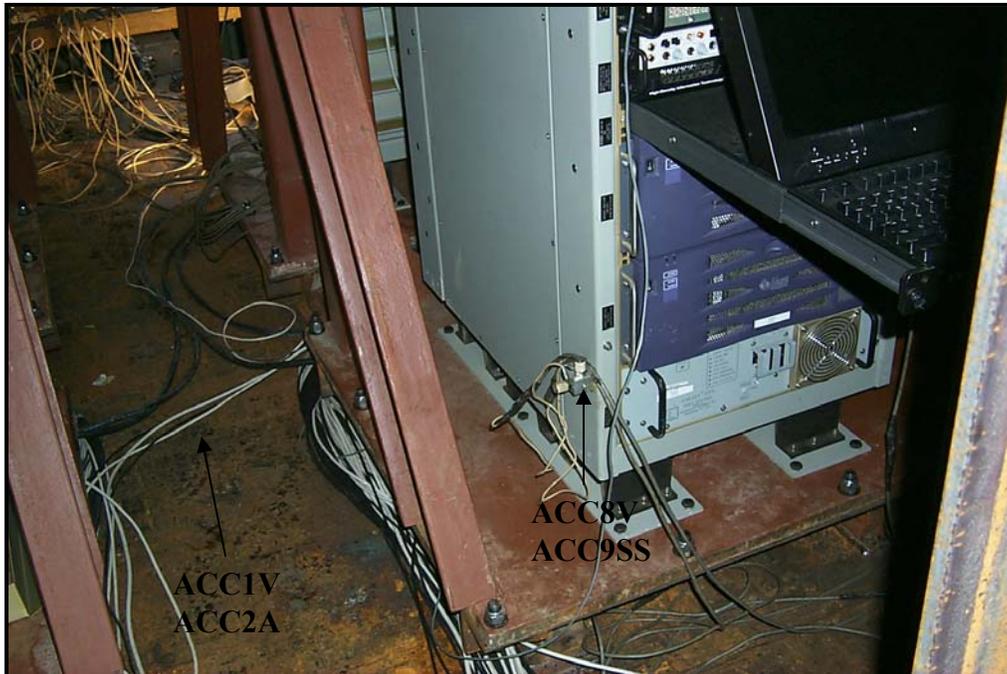
**Photograph No. 2. Overhead View of Initial Installation, Shot 4**

## 5.2 Test Instrumentation

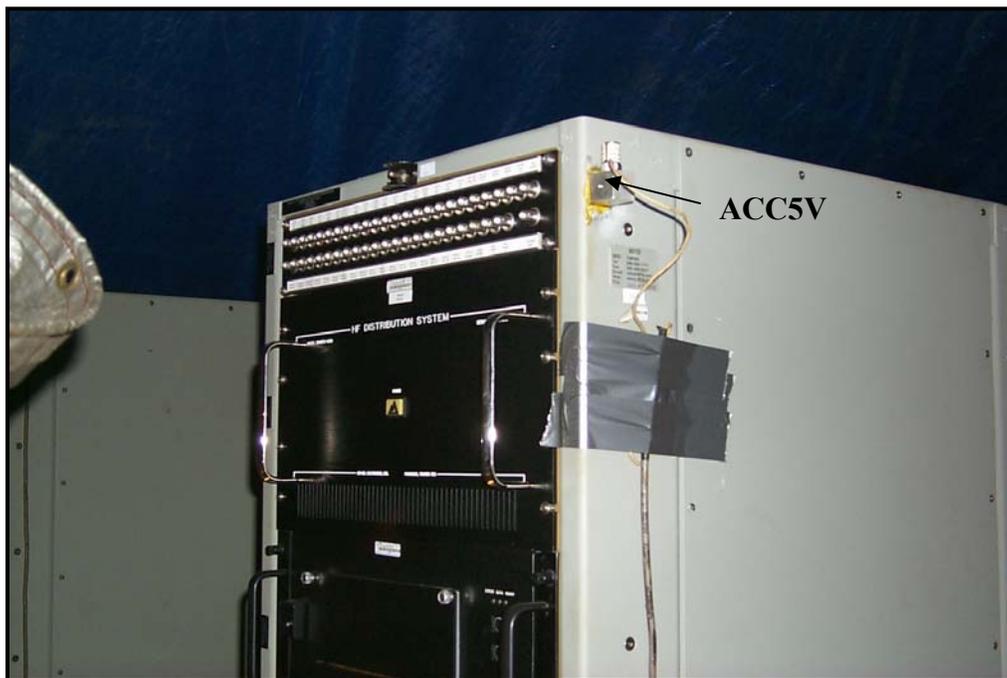
One velocity meter (VM), and nine accelerometers (ACC) were installed on the equipment and FSP during the test series to verify proper test geometries. One digital recording in Windows Media Player format was taken of each shot. Location of instrumentation listed in Table 4 can be seen in Photograph Nos. 3 through 7.

**Table 4. Test Instrumentation**

| <b>Item</b> | <b>Orientation</b> | <b>Location</b>                          |
|-------------|--------------------|--|
| VM1         | Vertical           | FSP Innerbottom, Blast Side              |
| A1V         | Vertical           | Center DSF                               |
| A2A         | Athwartship        | Center DSF                               |
| A5V         | Vertical           | Top Right Front of HF Receive Rack       |
| A6V         | Vertical           | Top Right Front of System Control Rack   |
| A7SS        | Side – Side        | Top Right Front of System Control Rack   |
| A8V         | Vertical           | Bottom Left Front of System Control Rack |
| A9SS        | Side - Side        | Bottom Left Front of System Control Rack |
| A10V        | Vertical           | Top Right Front of V/UHF Receive Rack    |
| A11V        | Vertical           | Top Right Front of CCOP Rack             |



**Photograph No. 3. Instrumentation Locations**



**Photograph No. 4. Instrumentation Location**



**Photograph No. 5. Instrumentation Location**



**Photograph No 6. Instrumentation Location**



**Photograph No. 7. Instrumentation Location**

## **6.0 TEST RESULTS**

### **6.1 Test Schedule**

**Table 5. Test Schedule Data**

| <b>Shot No./Test No.</b> | <b>Standoff</b> | <b>Date</b> | <b>Time (EST)</b> |
|--------------------------|-----------------|-------------|-------------------|
| 1/2                      | 30' Athwartship | 7-July-03   | 1222              |
| 2/3                      | 25' Athwartship | 8-July-03   | 1159              |
| 3/4                      | 20' Athwartship | 10-July-03  | 1231              |
| 4/4                      | 20' Athwartship | 11-July-03  | 1546              |

### **6.2 Survey Findings**

Pre- and post-test inspections and operational verifications were conducted before and after each shock test in accordance.

## 6.2.1 Survey Findings Prior to Testing

**6.2.1.1**      **Action:** The equipment under test (EUT) was inspected for proper installation and visual deficiencies.

**Observation 1:** No issues.

**Resolution 1:** N/A

**6.2.1.2**      **Action:** The EUT had operational tests performed

**Observation 1:** No issues.

**Resolution 1:** N/A

## 6.2.2 Survey Findings after Shot 1, MIL-S-901D, 30-foot Standoff

**6.2.2.1**      **Action:** The EUT was inspected for proper installation and visual deficiencies.

**Observation 1:** No issues.

**Resolution 1:** N/A

**6.2.2.2**      **Action:** The EUT had operational tests performed

**Observation 1:** The KVM and keyboard were not operational in the System Control Rack.

**Resolution 1:** The front cover was removed and power cycled to the KVM and keyboard. This corrected the KVM and keyboard issues.

**Observation 2:** The HF Receive, V/UHF Receive, and System Control Racks were not operational after the shock event.

**Resolution 2:** The reset button on the System Control Rack was pushed to cycle power. This did not correct any of the system failures. The AC power was cycled on both servers and full operation was restored in approximately 9 minutes 45 seconds.

**Observation 3:** No further issues.

**Resolution 3:** N/A

## 6.2.3 Survey Findings after Shot 2, MIL-S-901D, 25-foot Standoff

**6.2.3.1**      **Action:** The EUT was inspected for proper installation and visual deficiencies.

**Observation 1:** No issues.

**Resolution 1:** N/A

**6.2.3.2**      **Action:** The EUT had operational tests performed

**Observation 1:** The KVM and keyboard was not operational in the System Control Rack.

**Resolution 1:** The front cover was removed and power cycled to the KVM and keyboard. This corrected the KVM and keyboard issues.

**Observation 2:** The HF Receive, V/UHF Receive, and System Control Racks were not operational after the shock event.

**Resolution 2:** The reset button on the System Control Rack was pushed to cycle power. This did not correct any of the system failures. The AC power was cycled on both servers and full operation was restored in approximately 9 minutes 45 seconds.

**Observation 3:** No further issues.

**Resolution 3:** N/A

**NOTE 1:** The UPSs on only two of the four cabinets were operational during the first two shots.

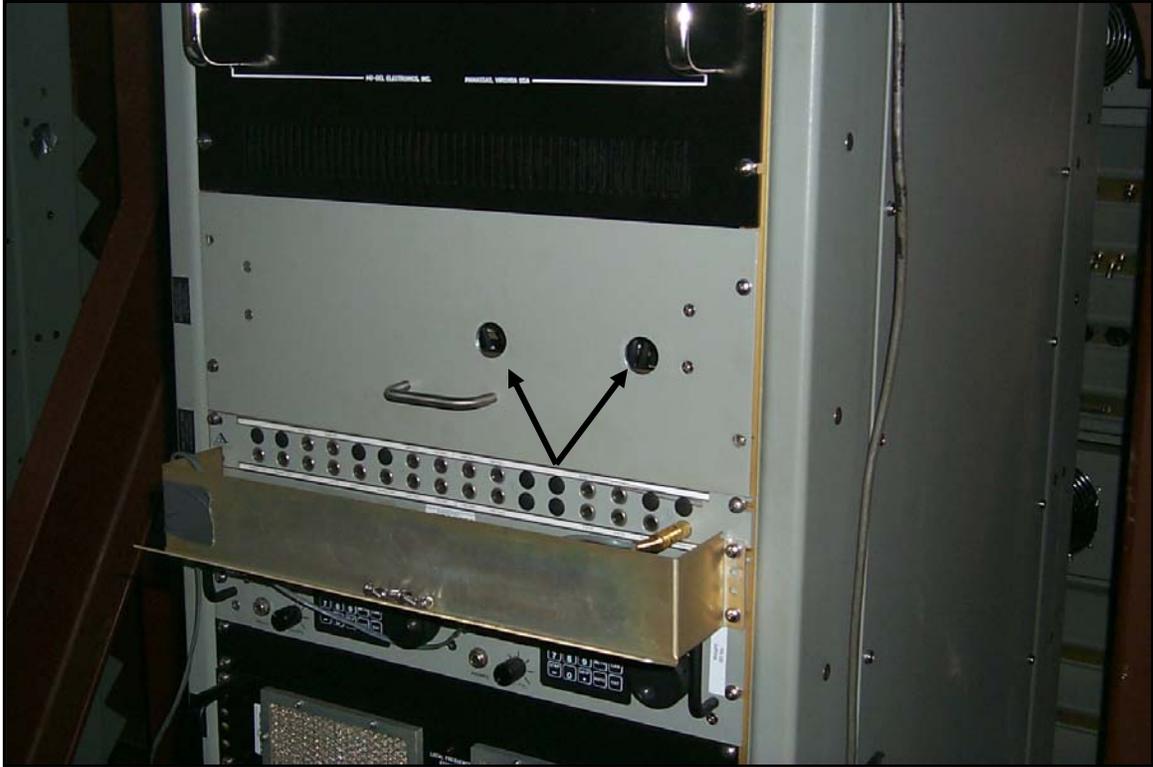


**Photograph No. 8. New UPS**

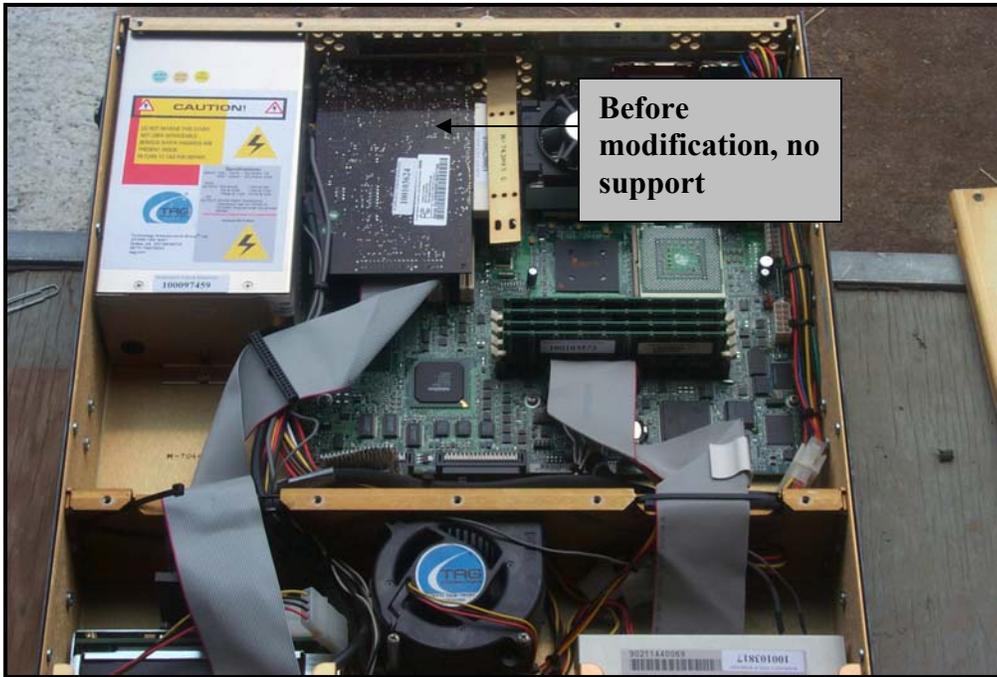


**Photograph No. 9. Location of New UPS in Racks**

**NOTE 2:** To enable the operator of the rack to reset the KVM, the cover was removed and modified, as seen in Photograph No. 10. Two 1-inch holes were drilled into the cover to allow for operator reset.



**Photograph No. 10. Modification to KVM Cover**



**Photograph No. 11. Before Modification**



**Photograph No. 12. After Modification**

#### **6.2.4 Survey Findings after Shot 3, MIL-S-901D, 20-foot Standoff,**

- 6.2.4.1 Action:** The EUT was inspected for proper installation and visual deficiencies.

**Observation 1:** No issues

**Resolution 1:** N/A

**6.2.4.2 Action:** The EUT had operational tests performed

**Observation 1:** The R-2412 Circuit Breaker in the V/UHF Receive Rack had tripped.

**Resolution 1:** The circuit breaker was reset.

**Observation 2:** The V/UHF Receive and System Control Racks were not operational after the shock event.

**Resolution 2:** The racks were not operational because the UPS in the V/UHF Rack was off-line. The UPS was reset but did not operate. The UPS was by-passed and all EUT were operational.

**Observation 3:** No further issues.

**Resolution 3:** N/A

**NOTE:** The EUT was rotated 90° after completion of the post-shot inspection. Photograph No. 4 shows the rotated test setup.

## **6.2.5 Survey Findings after Shot 4, MIL-S-901D, 20-foot Standoff,**

**6.2.5.1 Action:** The EUT was inspected for proper installation and visual deficiencies.

**Observation 1:** No issues.

**Resolution 1:** N/A

**6.2.5.2 Action:** The EUT had operational tests performed

**Observation 1:** The HF Receive Rack was not operating correctly.

**Resolution 1:** Rack was rebooted and operated correctly.

**Observation 2:** The V/UHF Receive Rack was not powered properly. UPS was not working.

**Resolution 2:** The on-site representative advised that a complete post-teardown inspection was required to troubleshoot the problems.

**Observation 3:** The System Control Rack Servers were not operating correctly.

**Resolution 3:** Servers were re-booted and operated correctly.

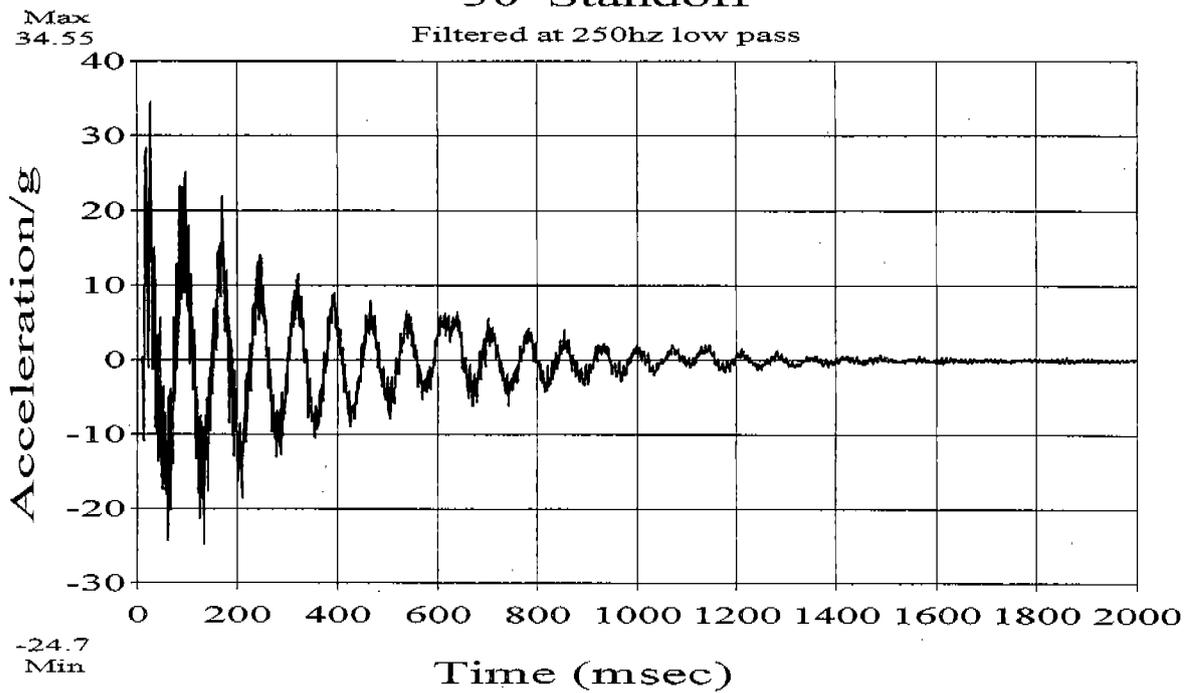
**Observation 4:** No further issues.

**Resolution 4:** This completed the test series.

**Appendix A**  
**Test Instrumentation Data**

# Shot1\_Test2\_J655

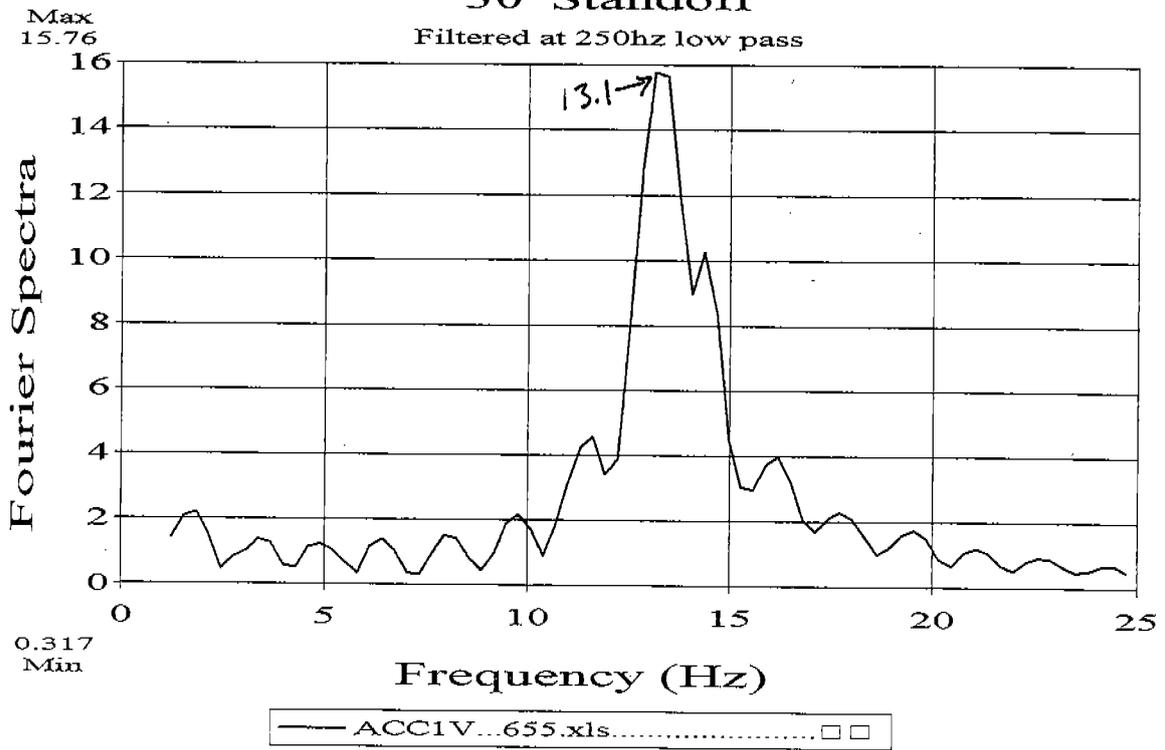
30' Standoff



— ACC1V...655.xls..... □ □

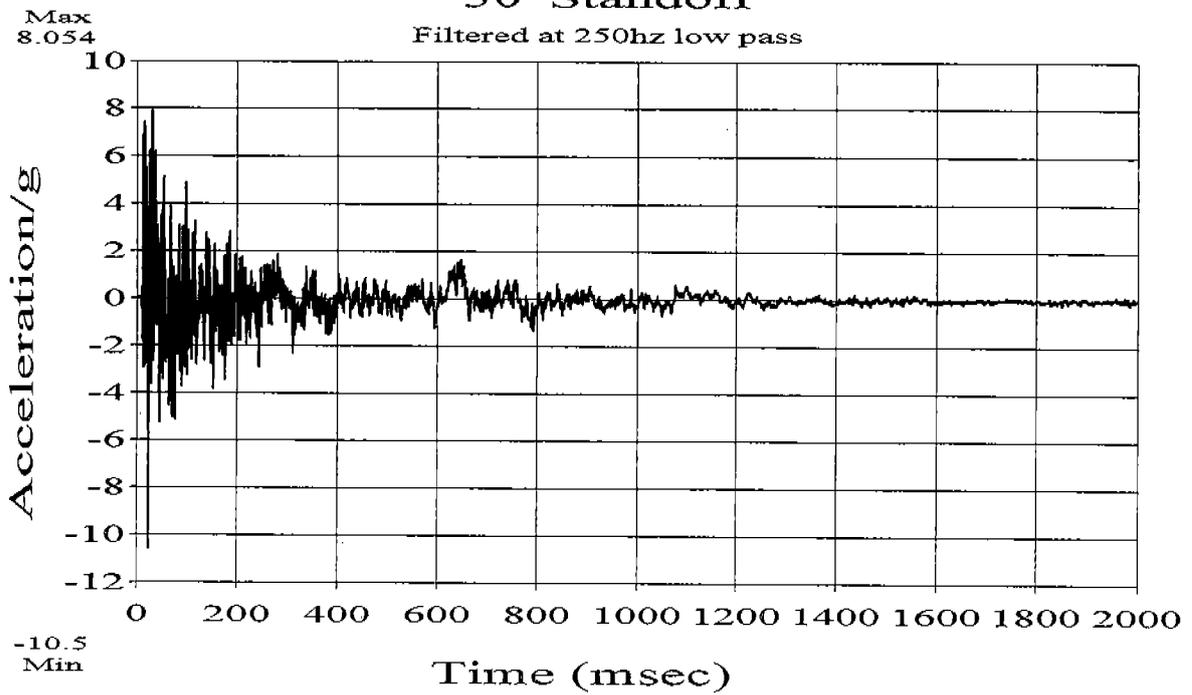
# Shot1\_Test2\_J655

30' Standoff



# Shot1\_Test2\_J655

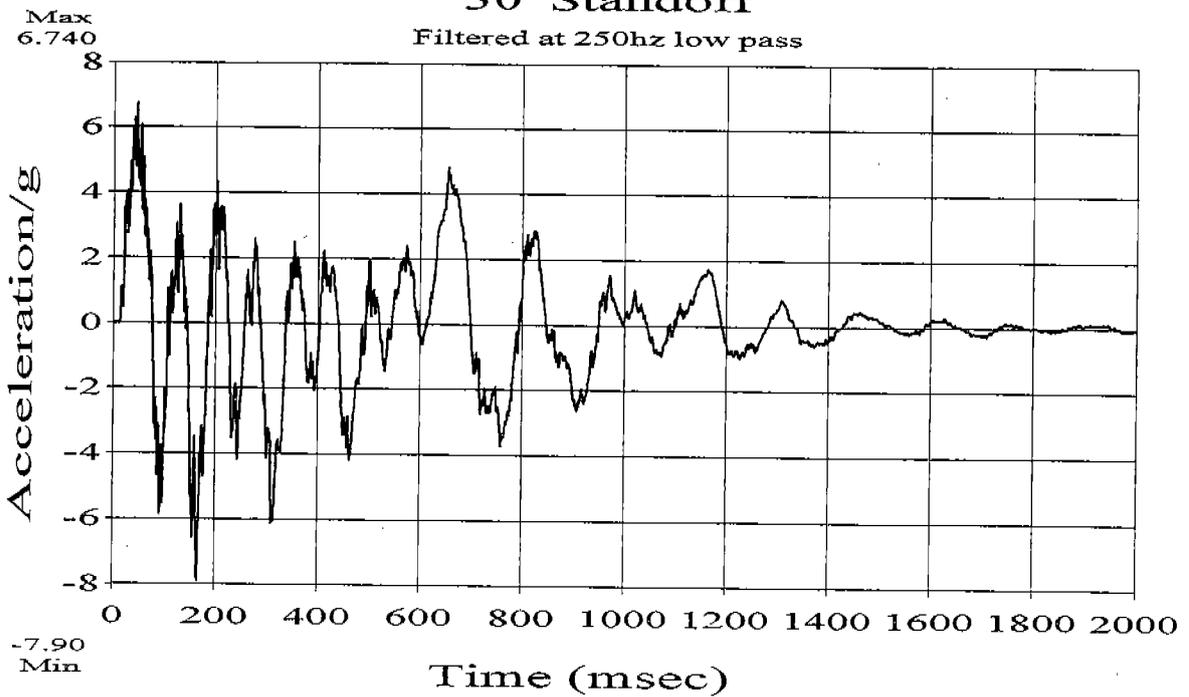
30' Standoff



— ACC2A...655.xls.....□□

# Shot1\_Test2\_J655

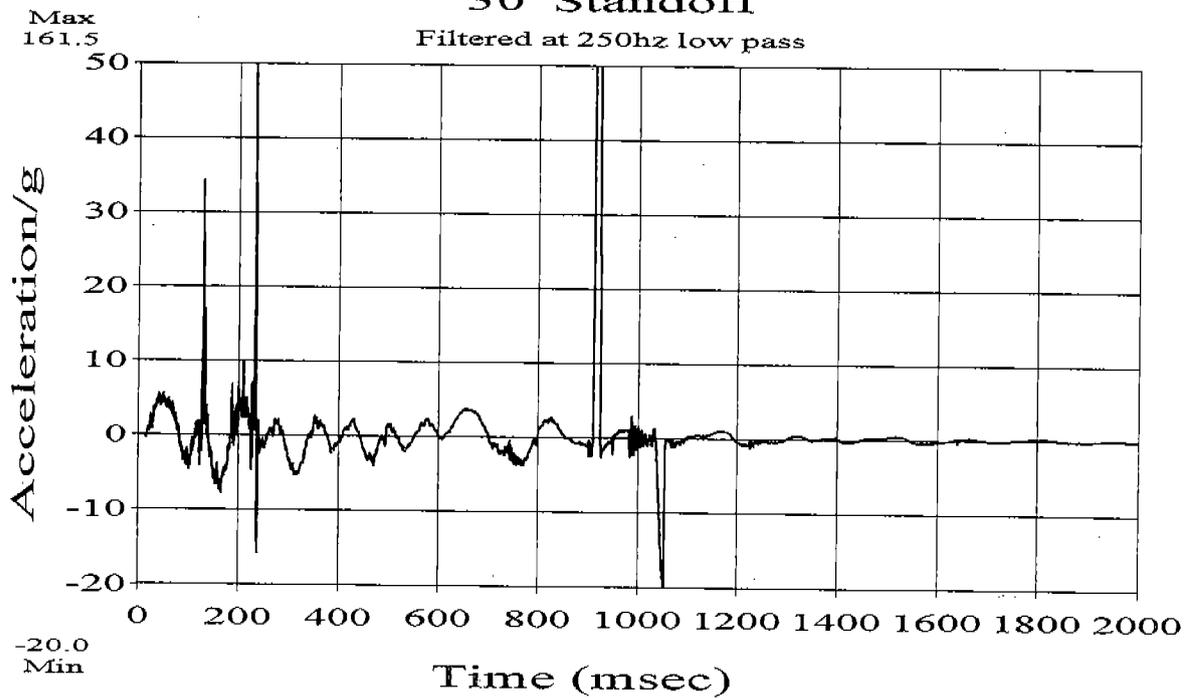
30' Standoff



— ACC5V...655.xls.....□□

# Shot1\_Test2\_J655

30' Standoff



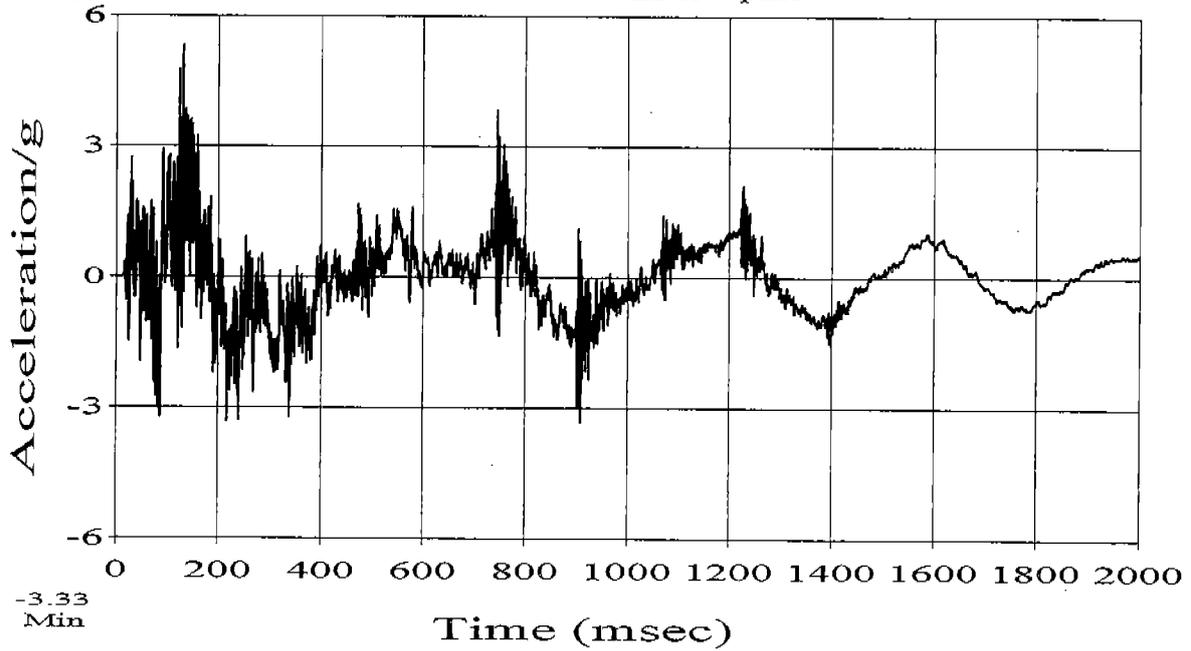
— ACC6V...655.xls..... □ □

# Shot1\_Test2\_J655

30' Standoff

Max  
5.318

Filtered at 250hz low pass



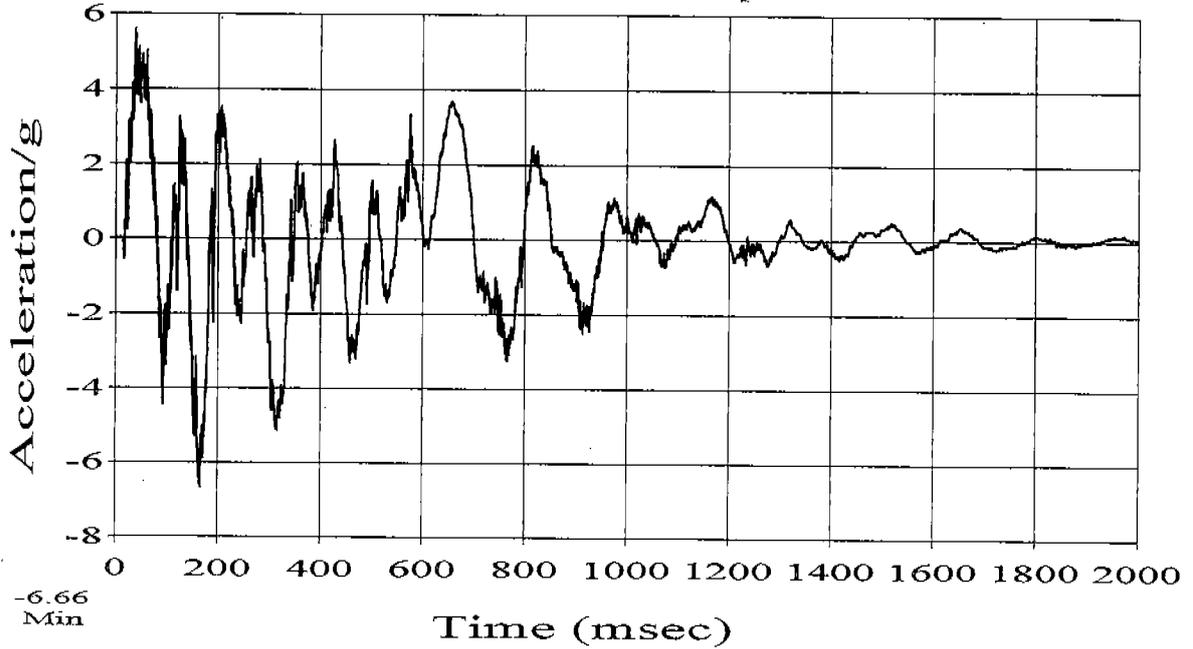
— ACC7SS...655.xls..... □ □

# Shot1\_Test2\_J655

30' Standoff

Max  
5.591

Filtered at 250hz low pass



-6.66  
Min

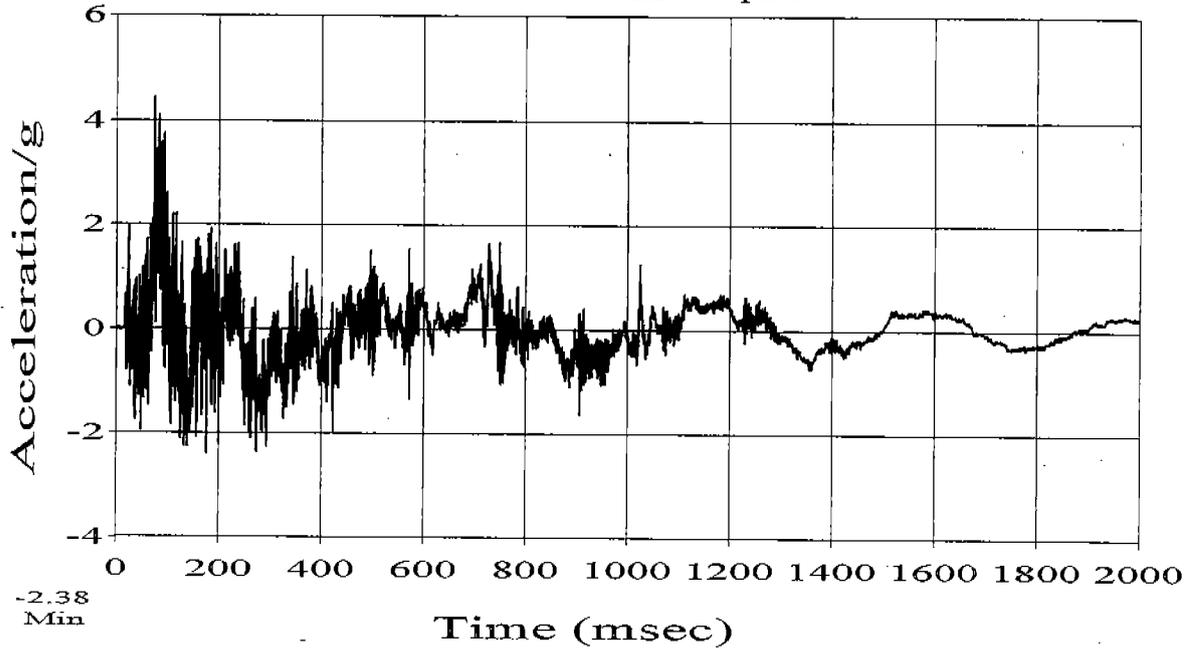
— ACC8V...655.xls..... □ □

# Shot1\_Test2\_J655

30' Standoff

Max  
4.444

Filtered at 250hz low pass

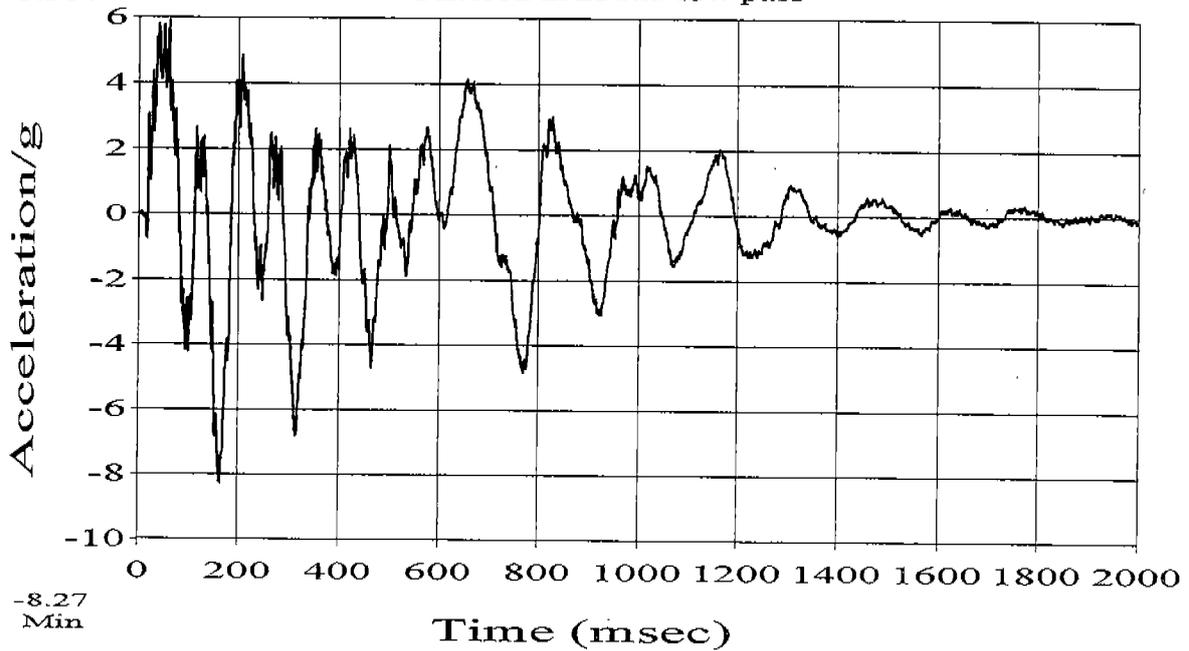


# Shot1\_Test2\_J655

30' Standoff

Max  
5.904

Filtered at 250hz low pass



-8.27  
Min

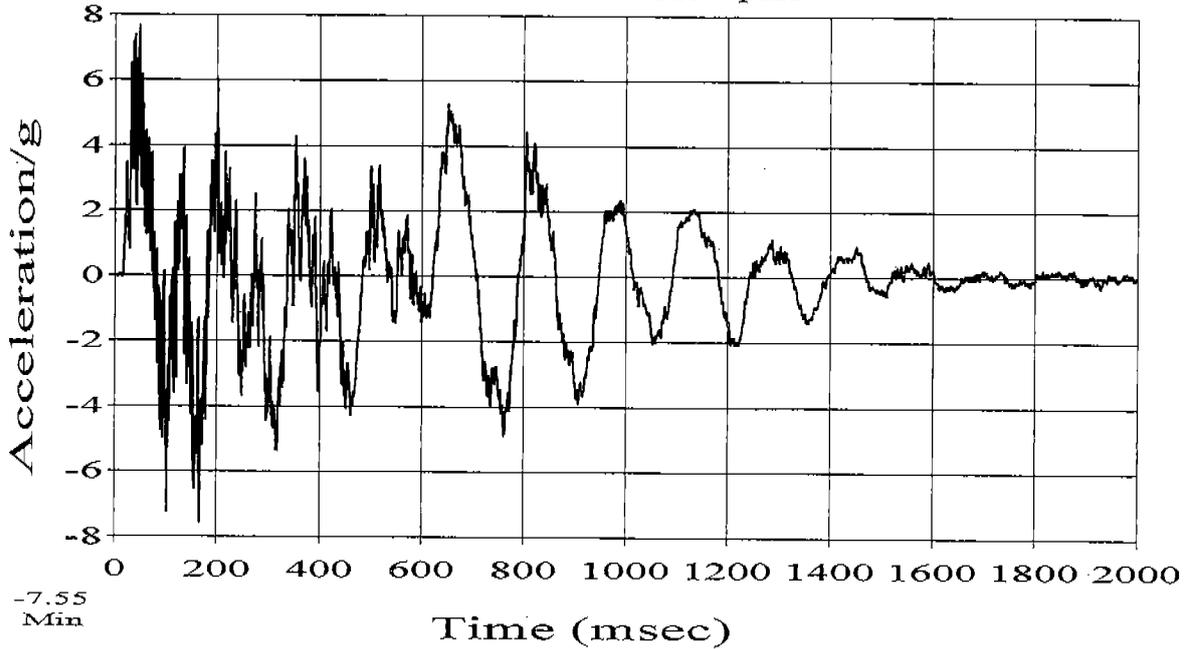
— ACC10V...655.xls.....□□

# Shot1\_Test2\_J655

30' Standoff

Max  
7.673

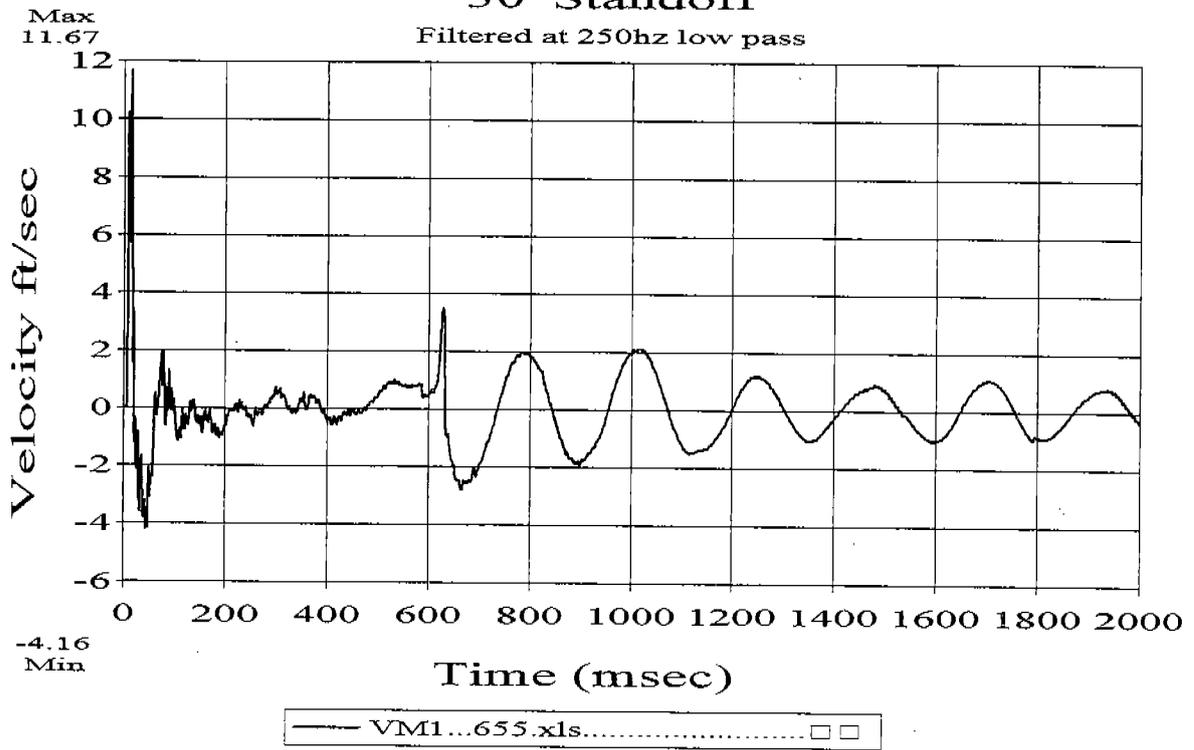
Filtered at 250hz low pass



— ACC11V...655.xls.....□□

# Shot1\_Test2\_J655

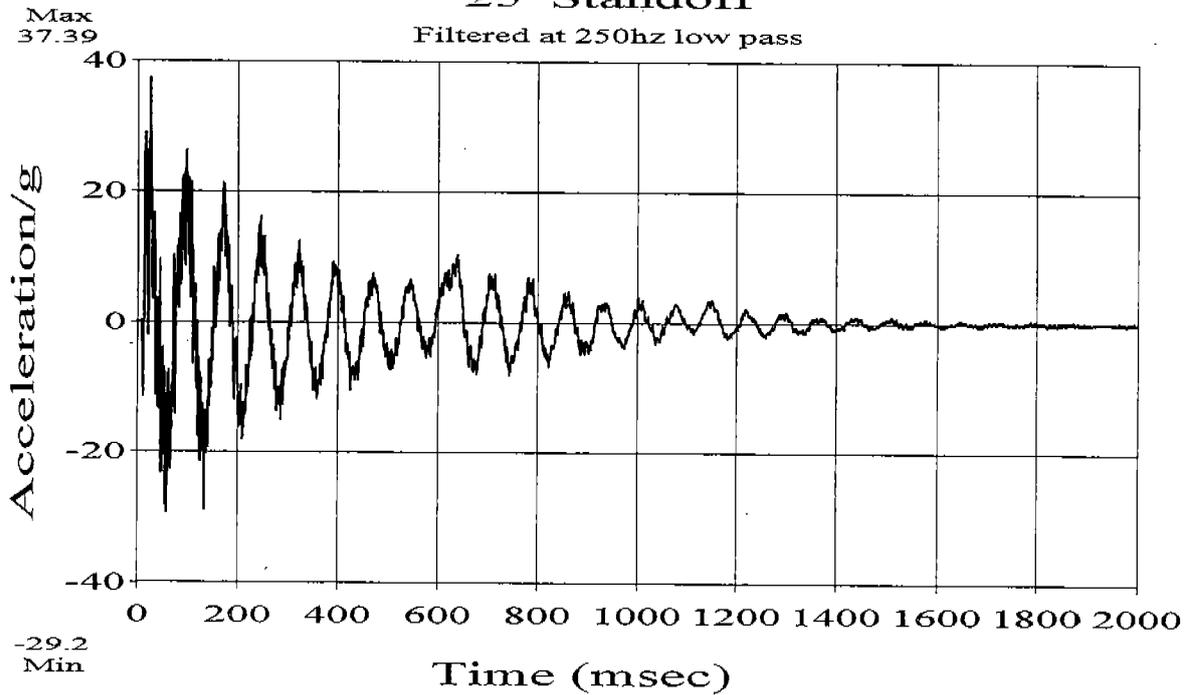
30' Standoff



# Shot2\_Test3\_J655

25' Standoff

Filtered at 250hz low pass

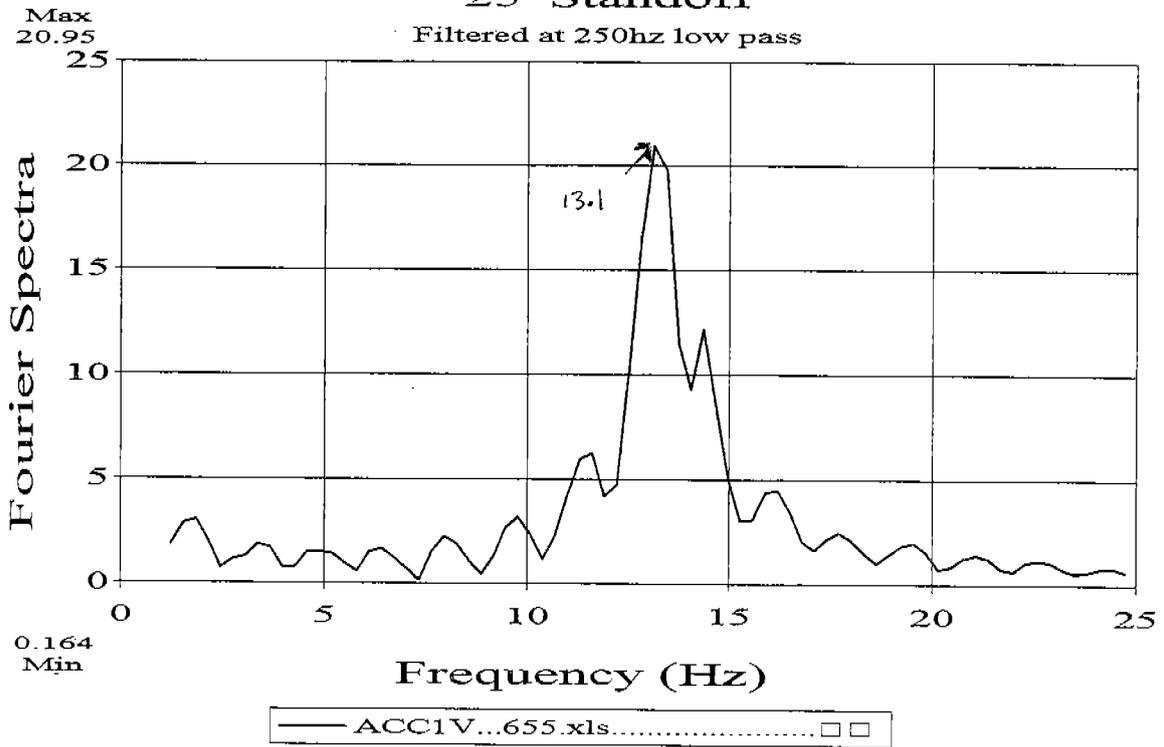


— ACC1V...655.xls..... □ □

# Shot2\_Test3\_J655

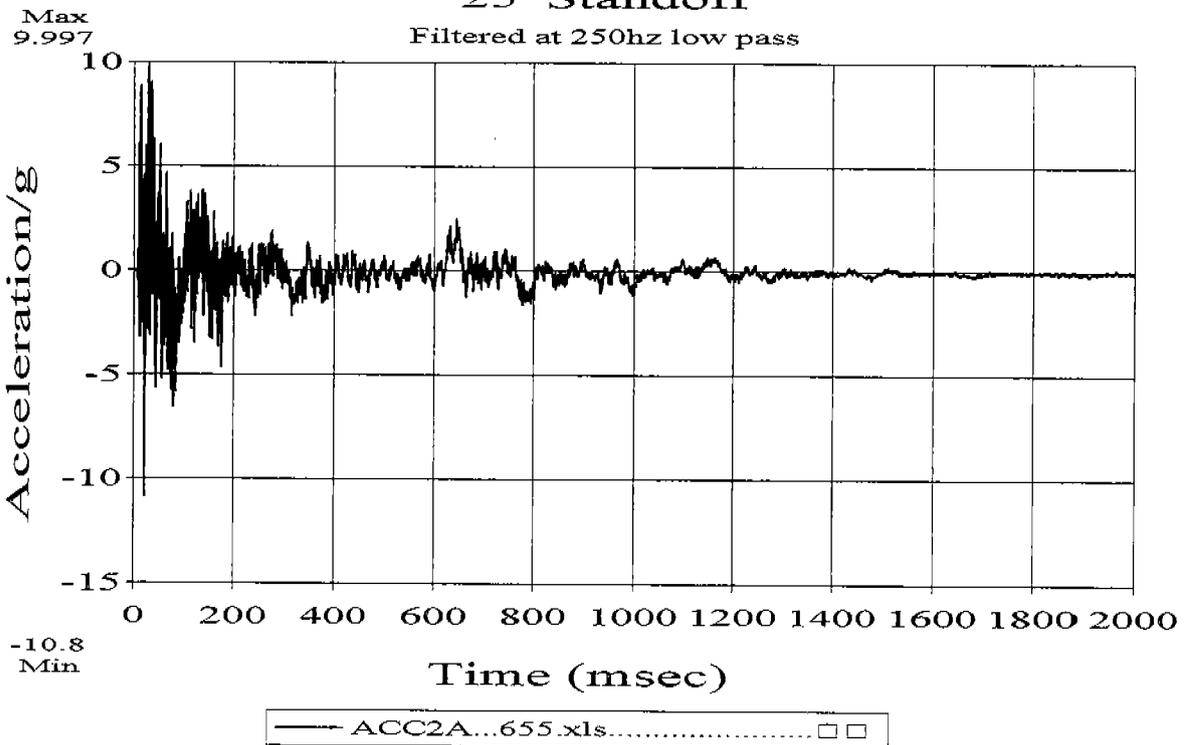
25' Standoff

Filtered at 250hz low pass



# Shot2\_Test3\_J655

25' Standoff

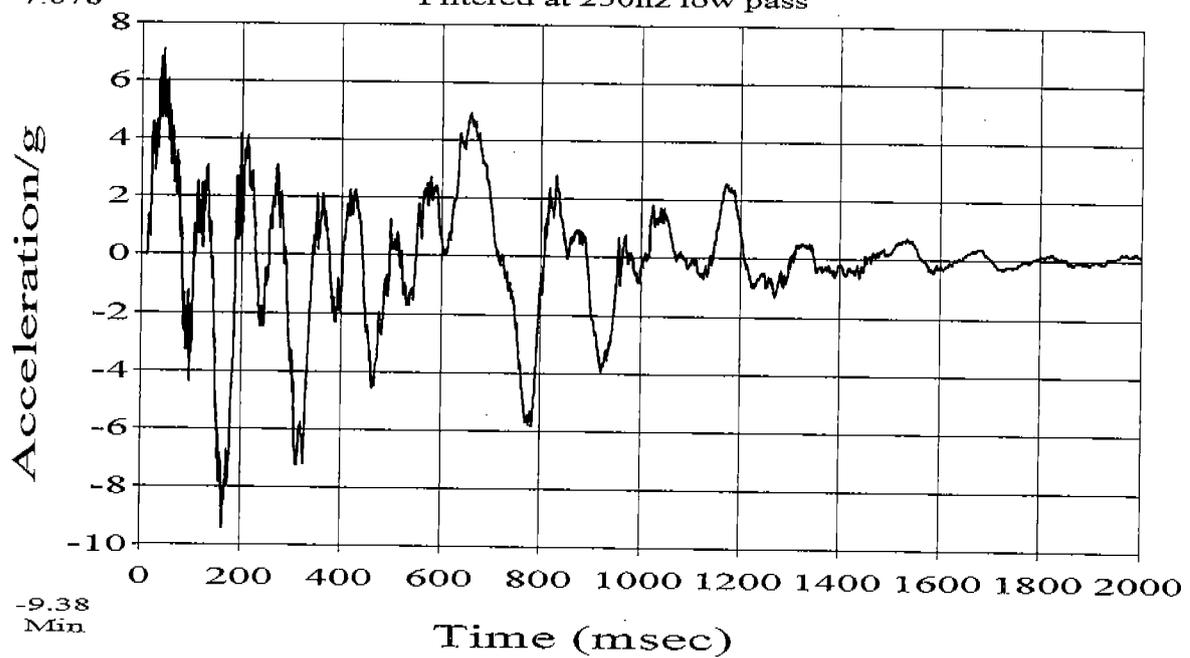


# Shot2\_Test3\_J655

25' Standoff

Max  
7.078

Filtered at 250hz low pass

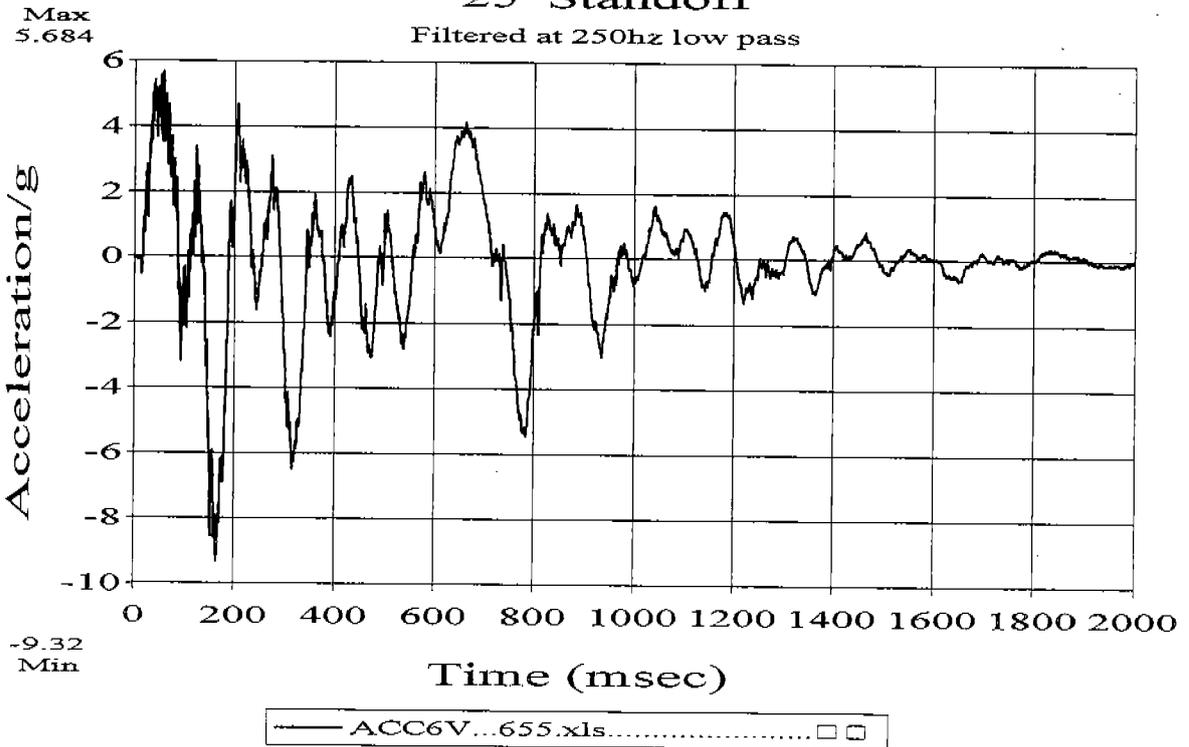


Min  
-9.38

— ACC5V...655.xls..... □ □

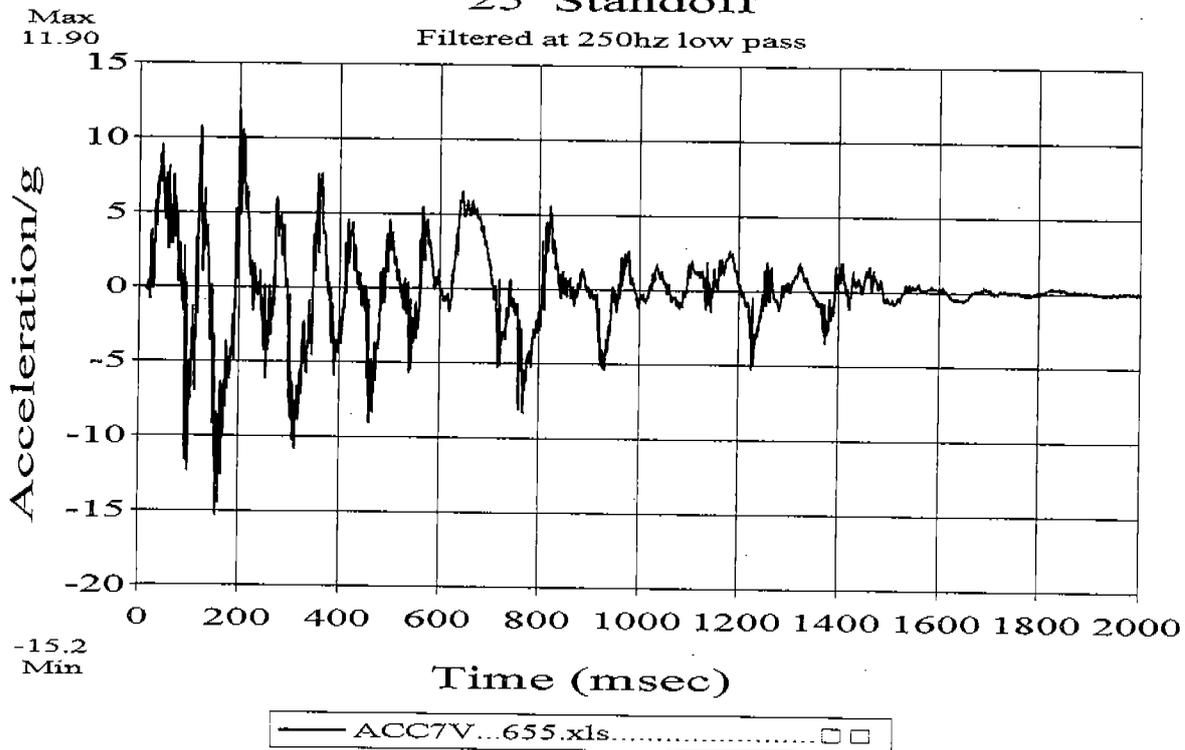
# Shot2\_Test3\_J655

## 25' Standoff



# Shot2\_Test3\_J655

25' Standoff

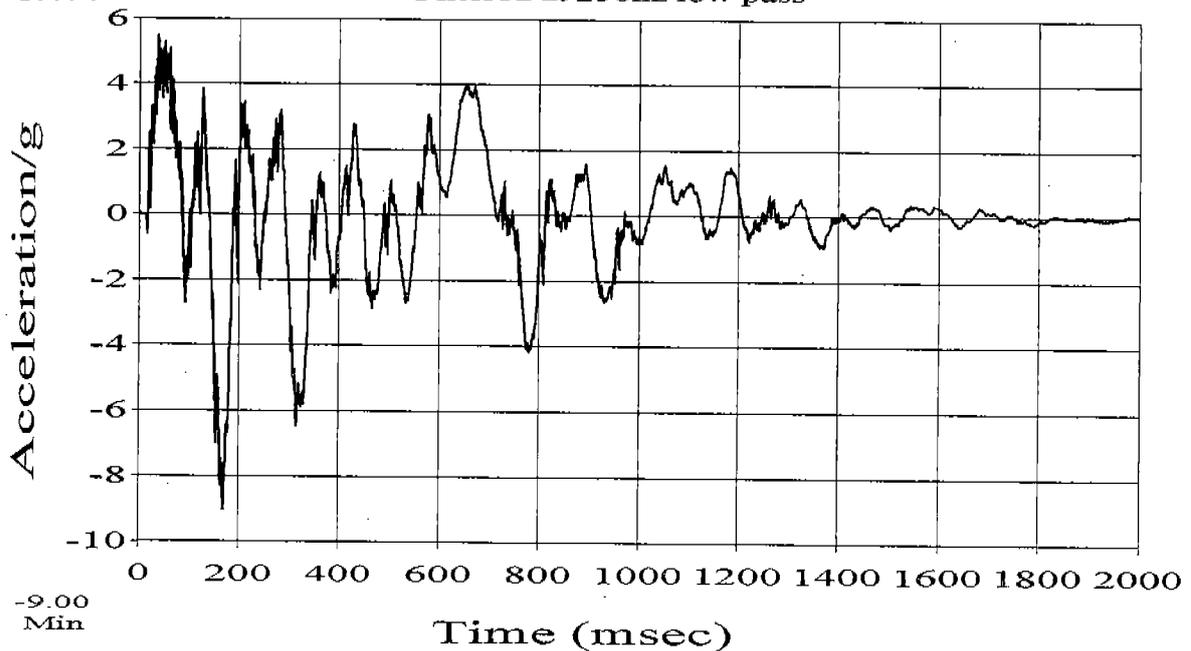


# Shot2\_Test3\_J655

25' Standoff

Max  
5.474

Filtered at 250hz low pass

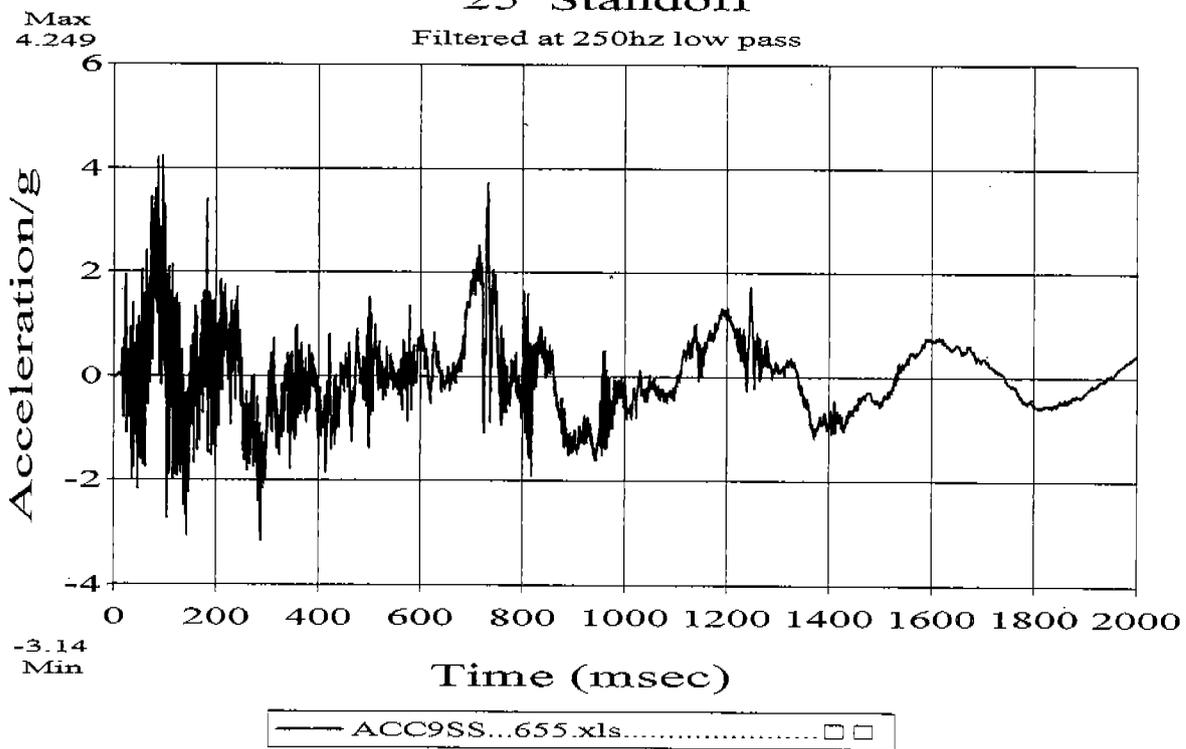


-9.00  
Min

— ACC8V...655.xls..... □ □

# Shot2\_Test3\_J655

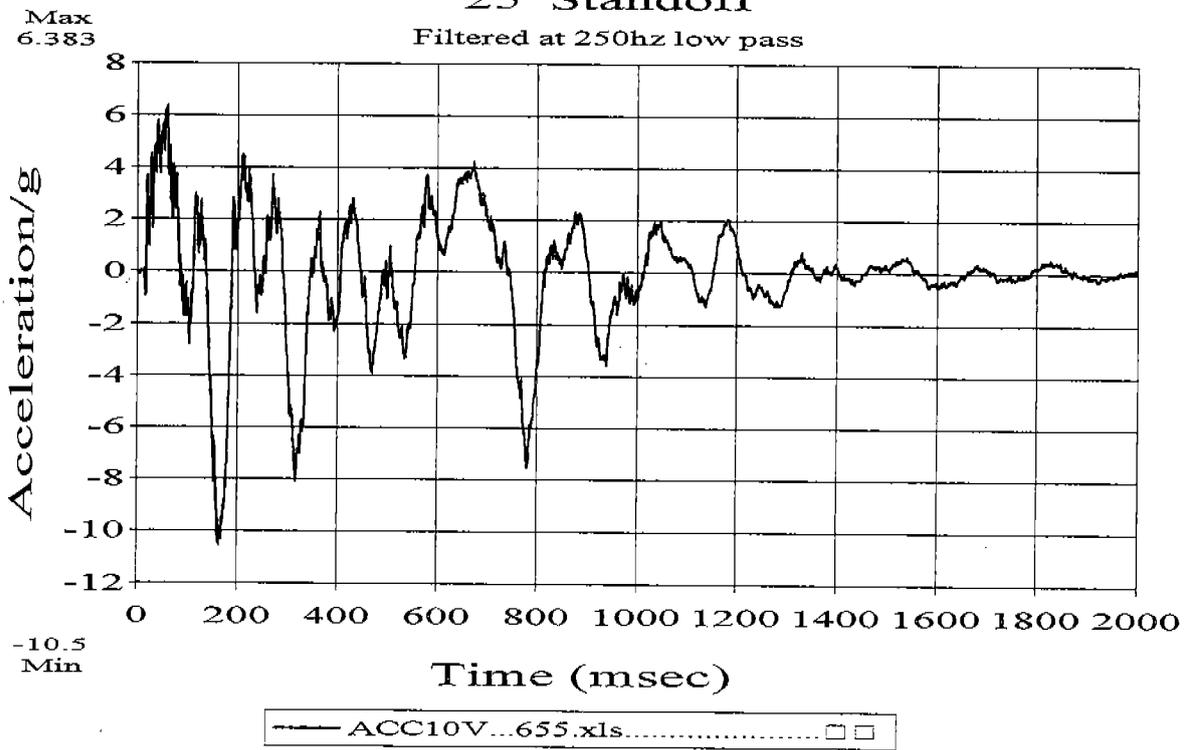
25' Standoff



# Shot2\_Test3\_J655

25' Standoff

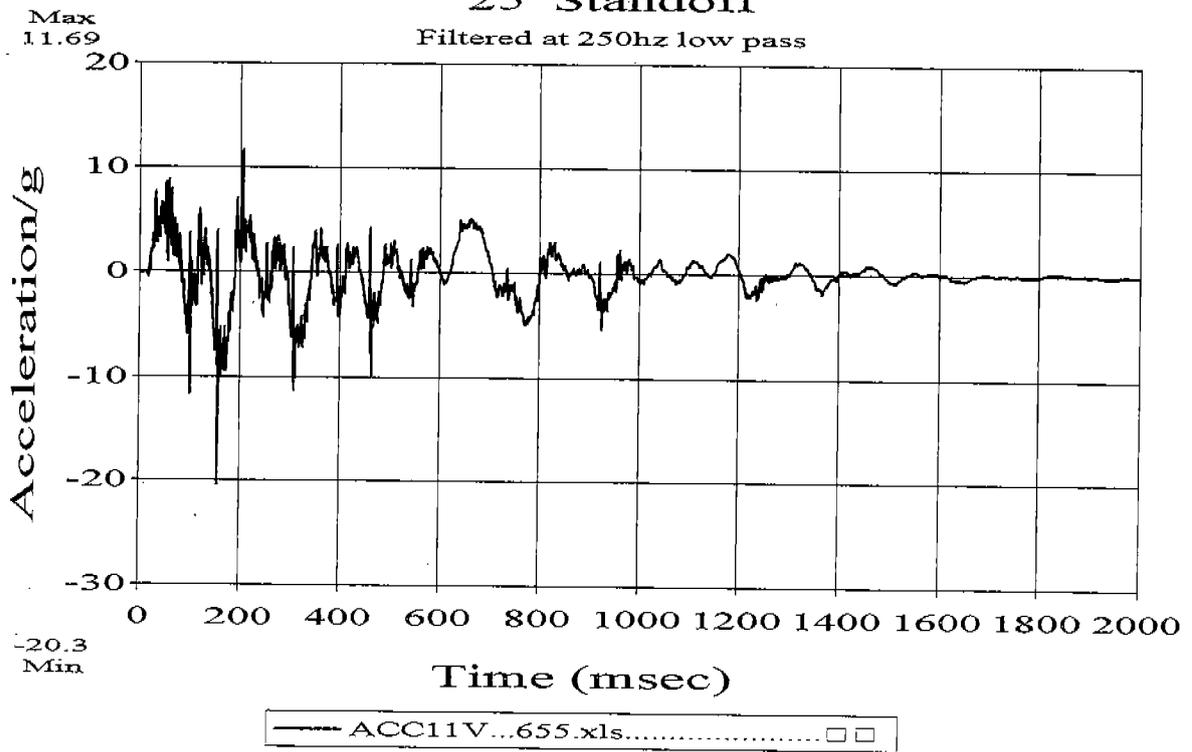
Filtered at 250hz low pass



# Shot2\_Test3\_J655

25' Standoff

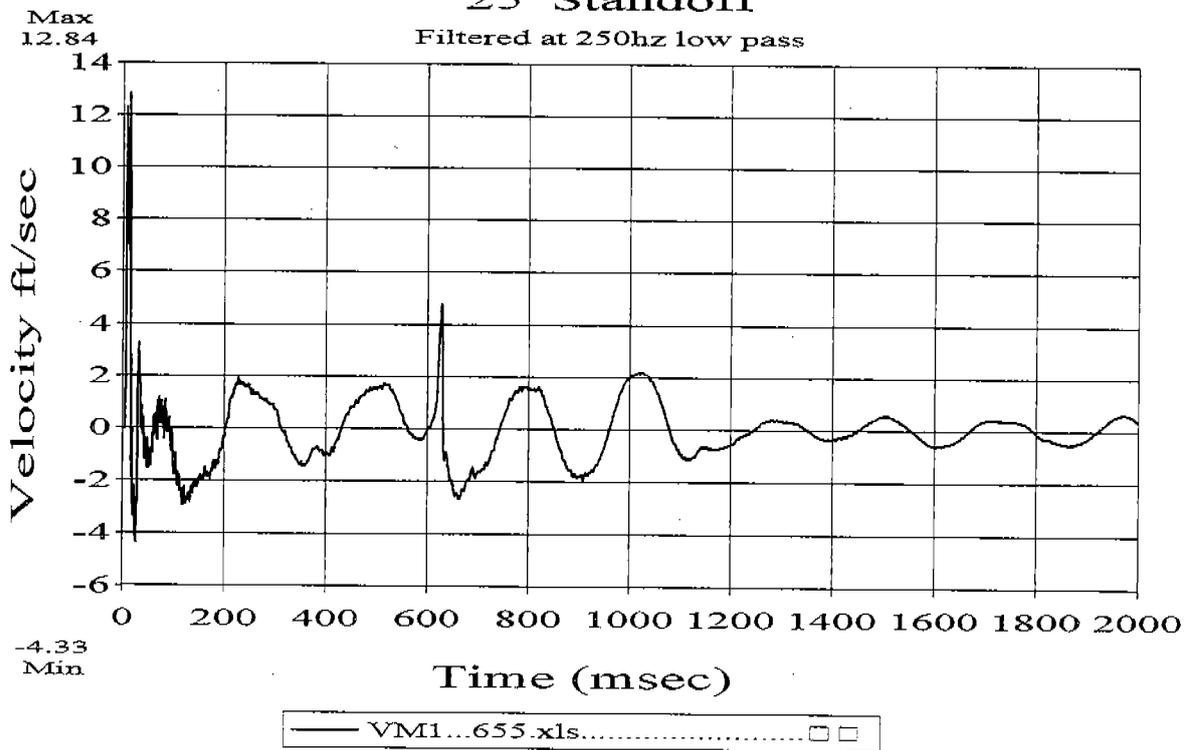
Filtered at 250hz low pass



# Shot2\_Test3\_J655

25' Standoff

Filtered at 250hz low pass

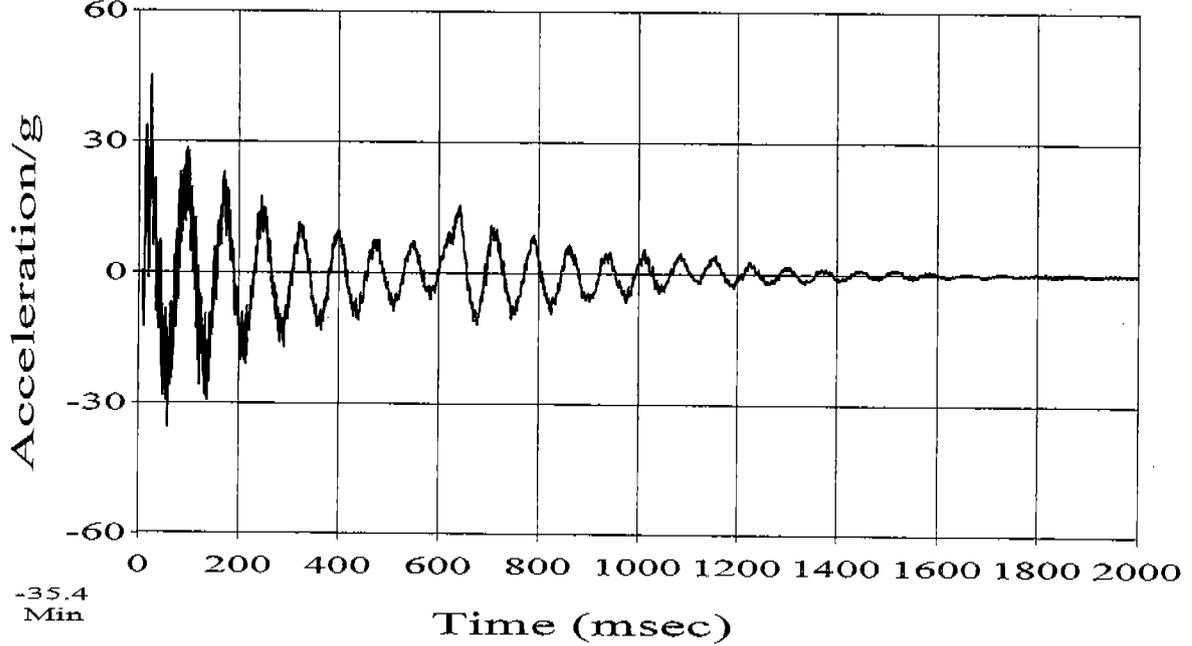


# Shot3\_Test4\_J655

20' Standoff

Max  
45.13

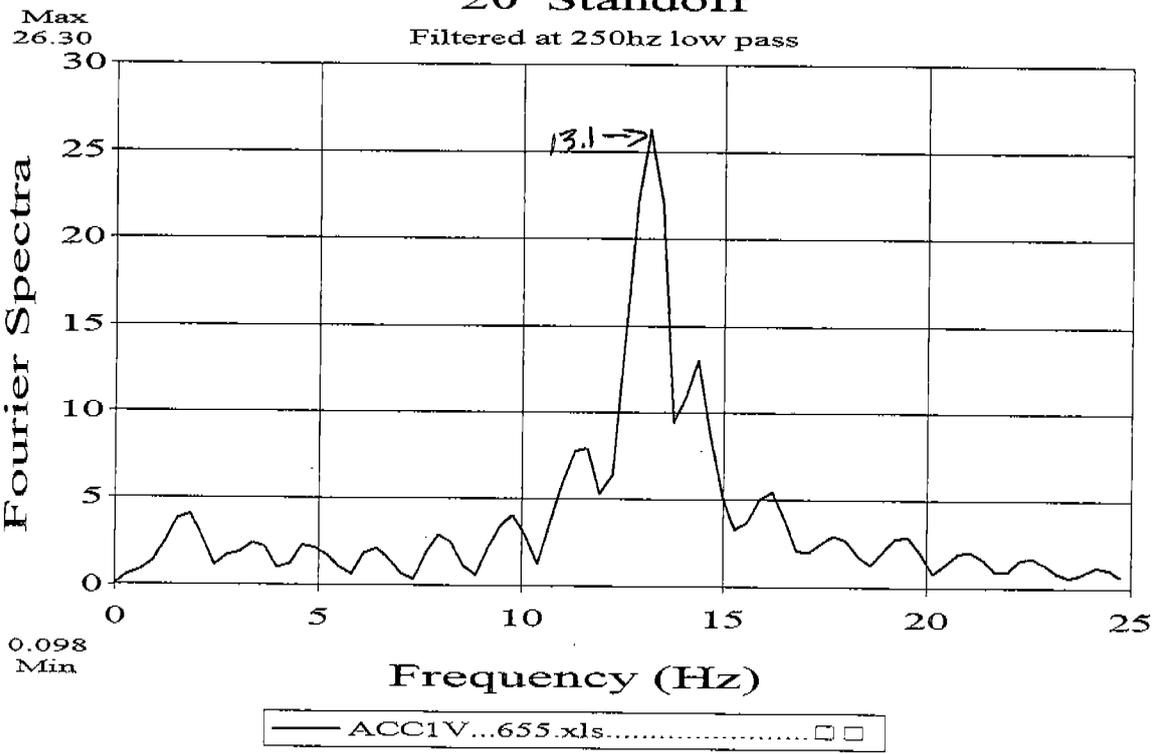
Filtered at 250hz low pass



— ACC1V...655.xls..... □ □

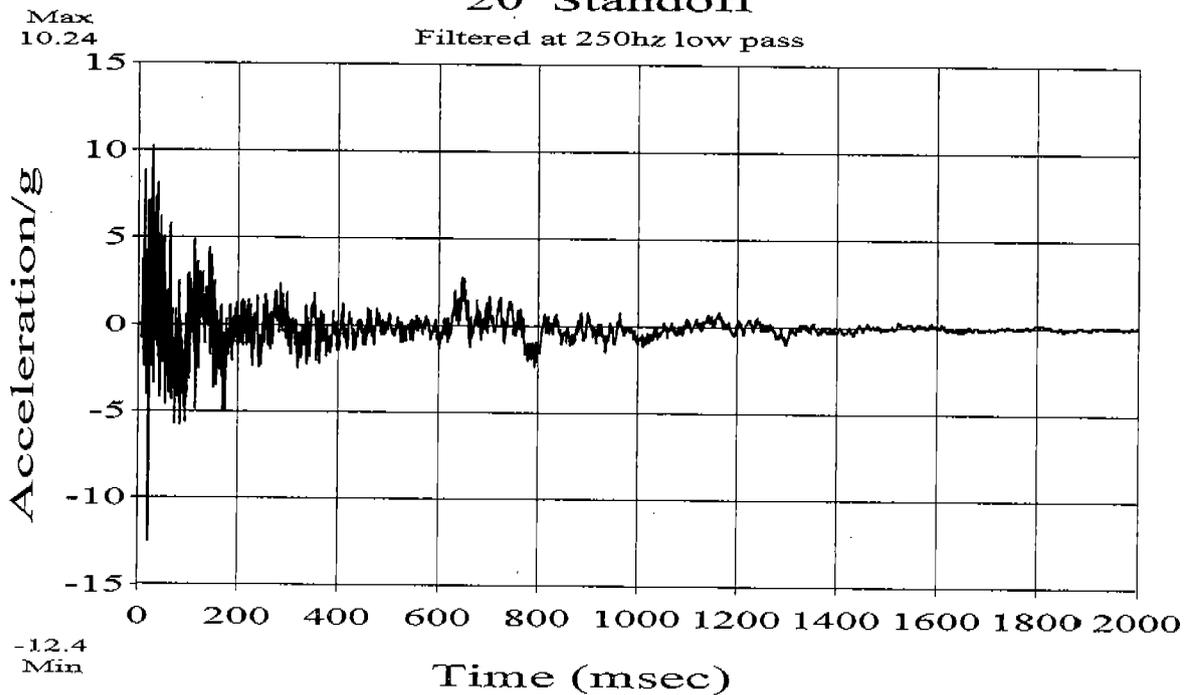
# Shot3\_Test4\_J655

20' Standoff



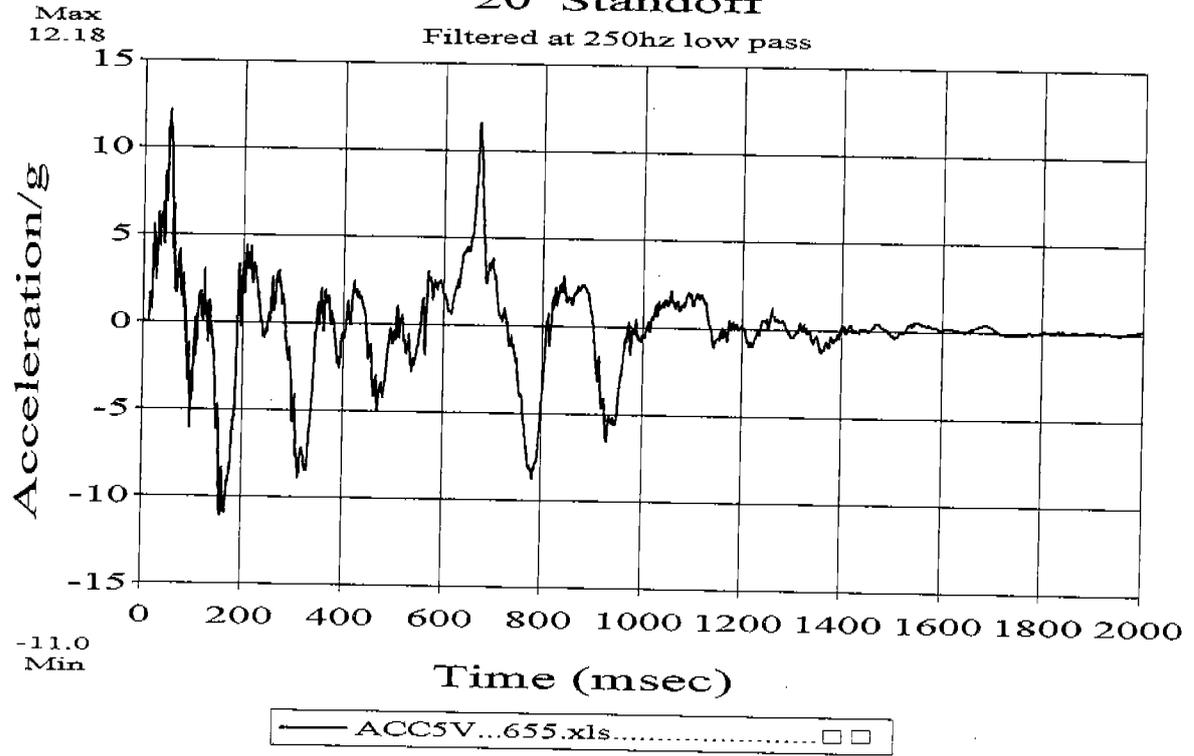
# Shot3\_Test4\_J655

20' Standoff



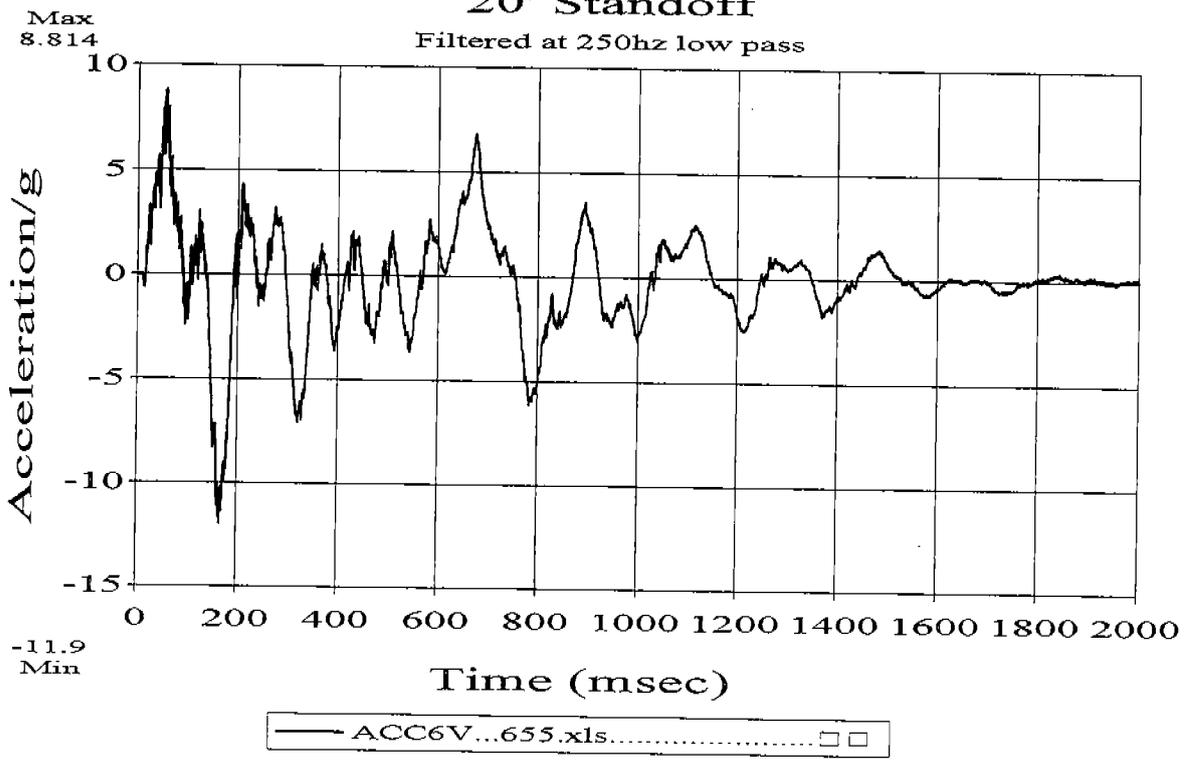
# Shot3\_Test4\_J655

20' Standoff



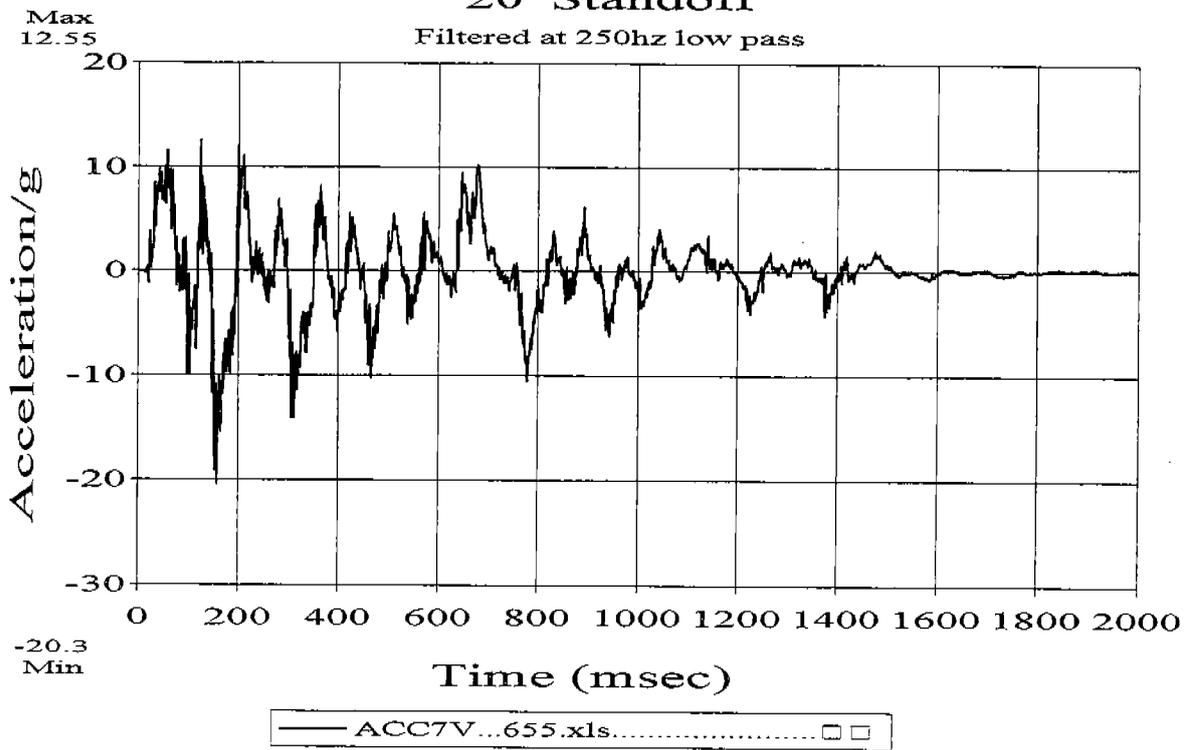
# Shot3\_Test4\_J655

20' Standoff



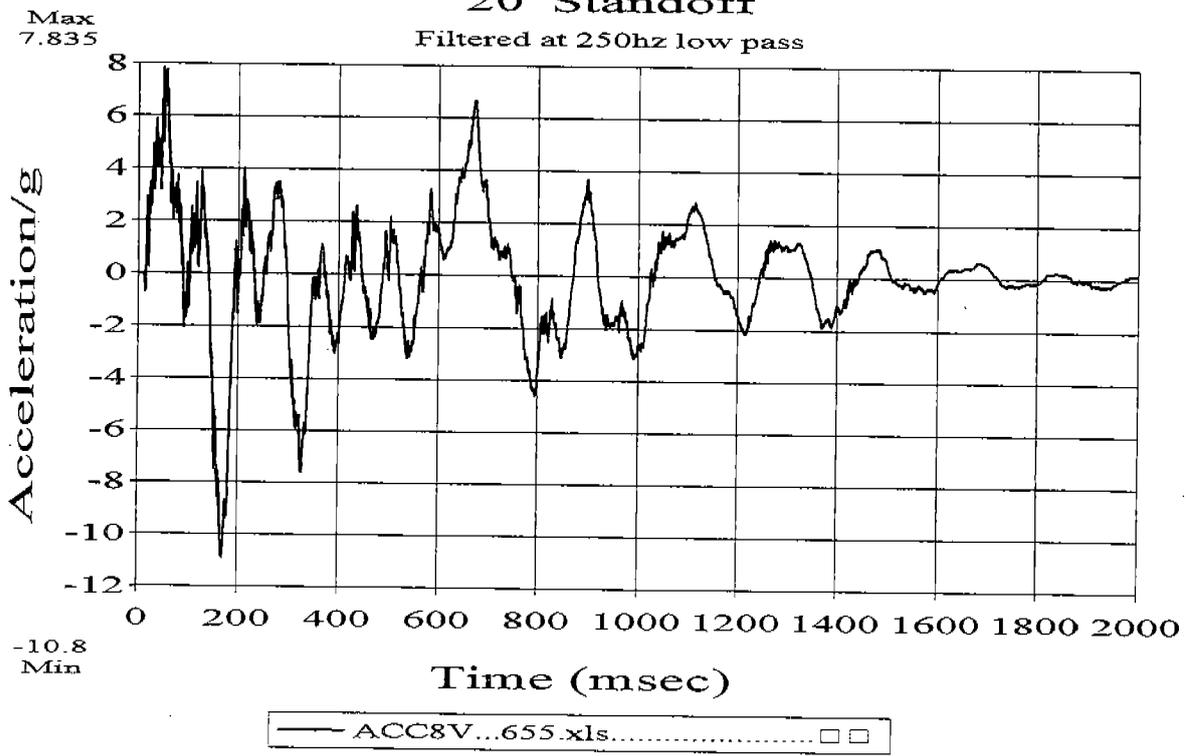
# Shot3\_Test4\_J655

20' Standoff



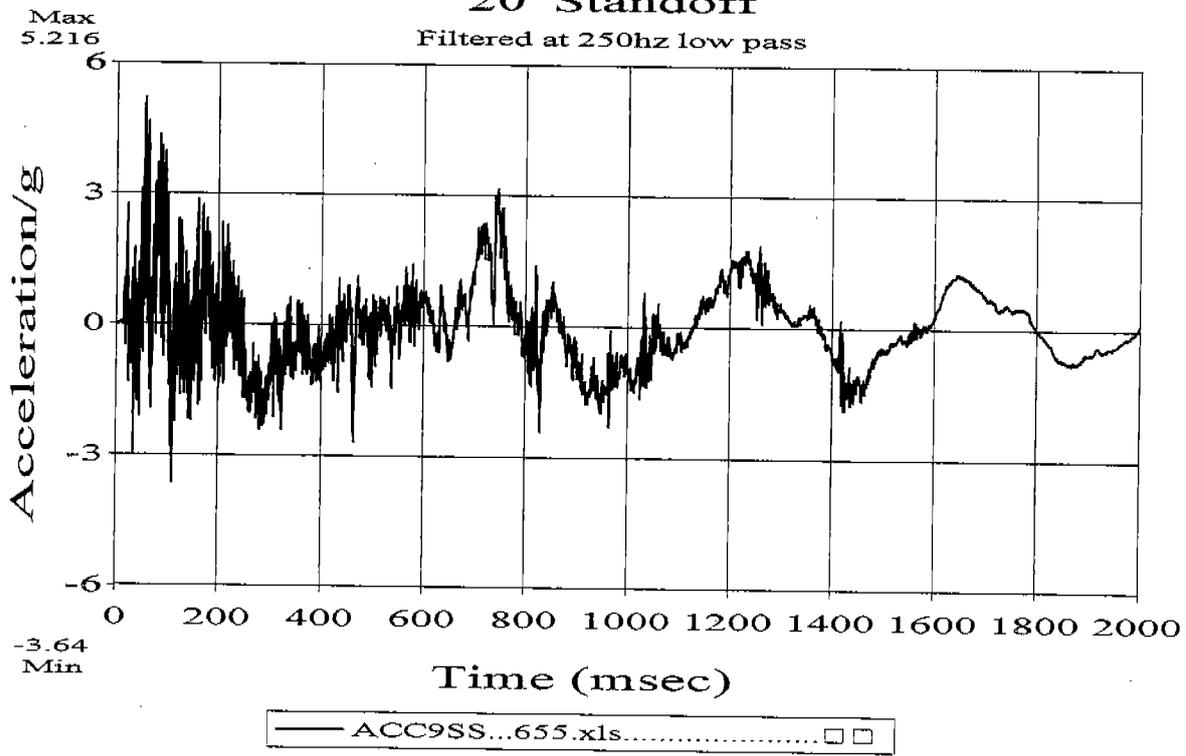
# Shot3\_Test4\_J655

20' Standoff



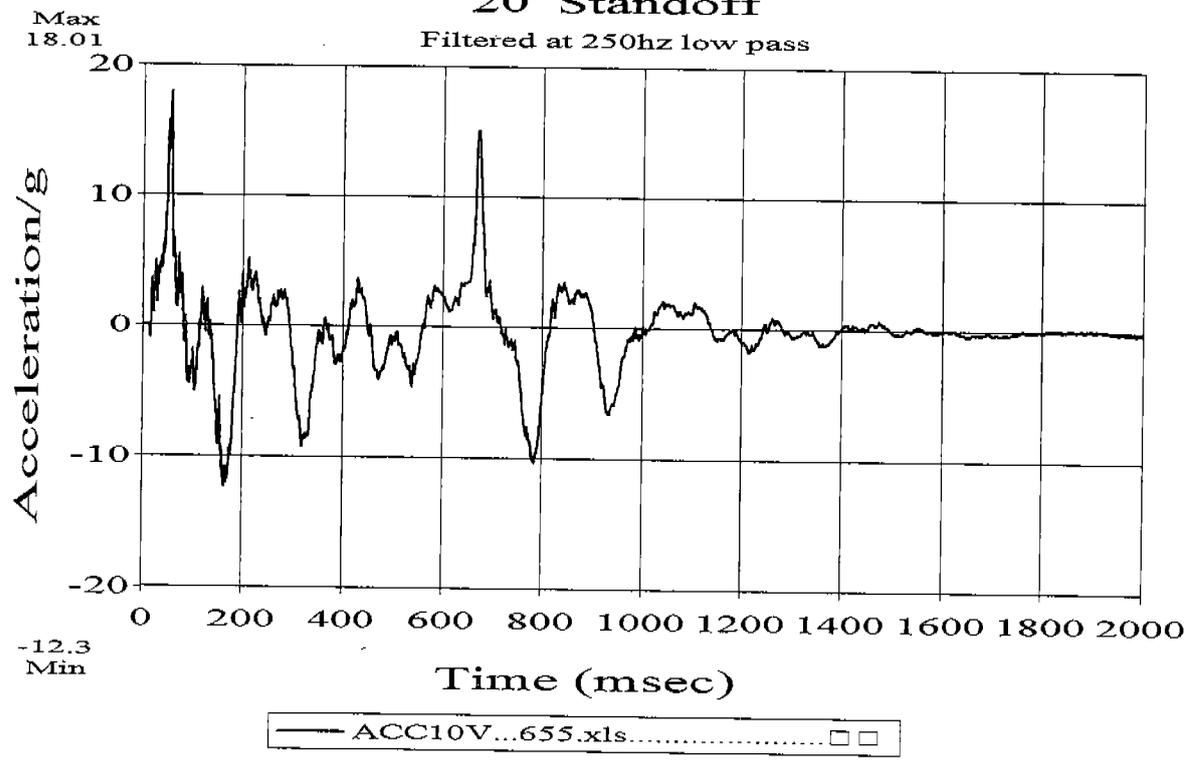
# Shot3\_Test4\_J655

20' Standoff



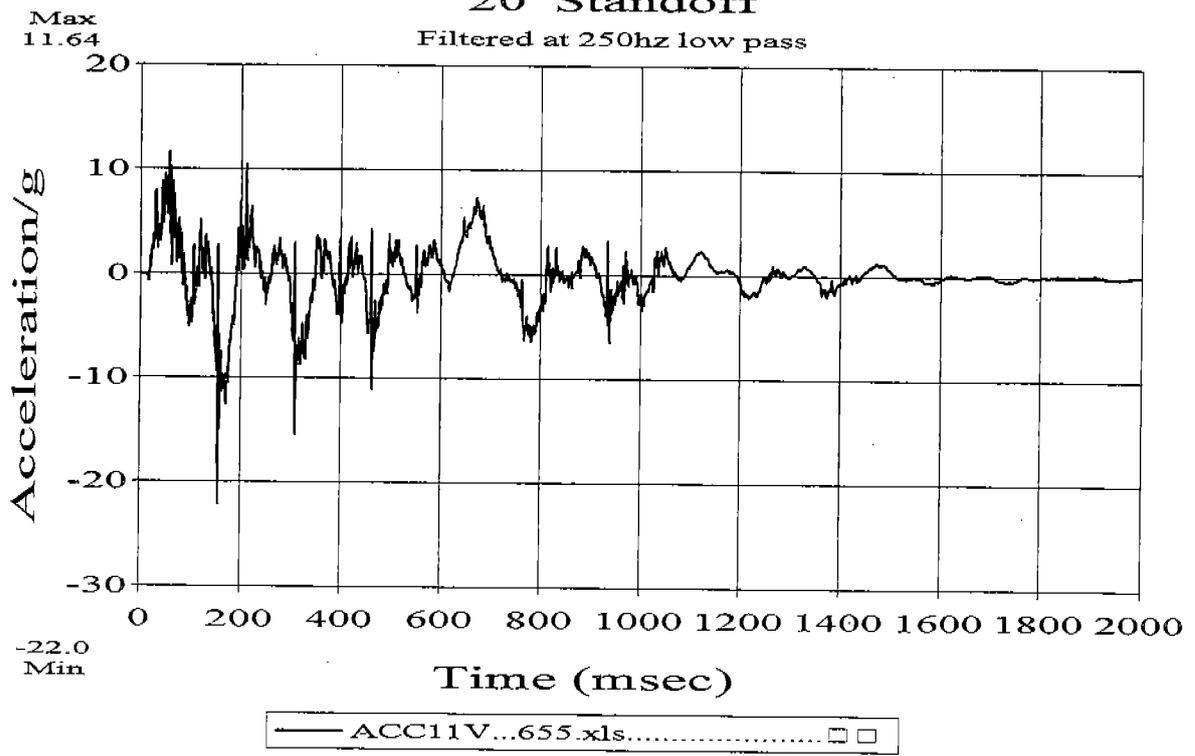
# Shot3\_Test4\_J655

20' Standoff



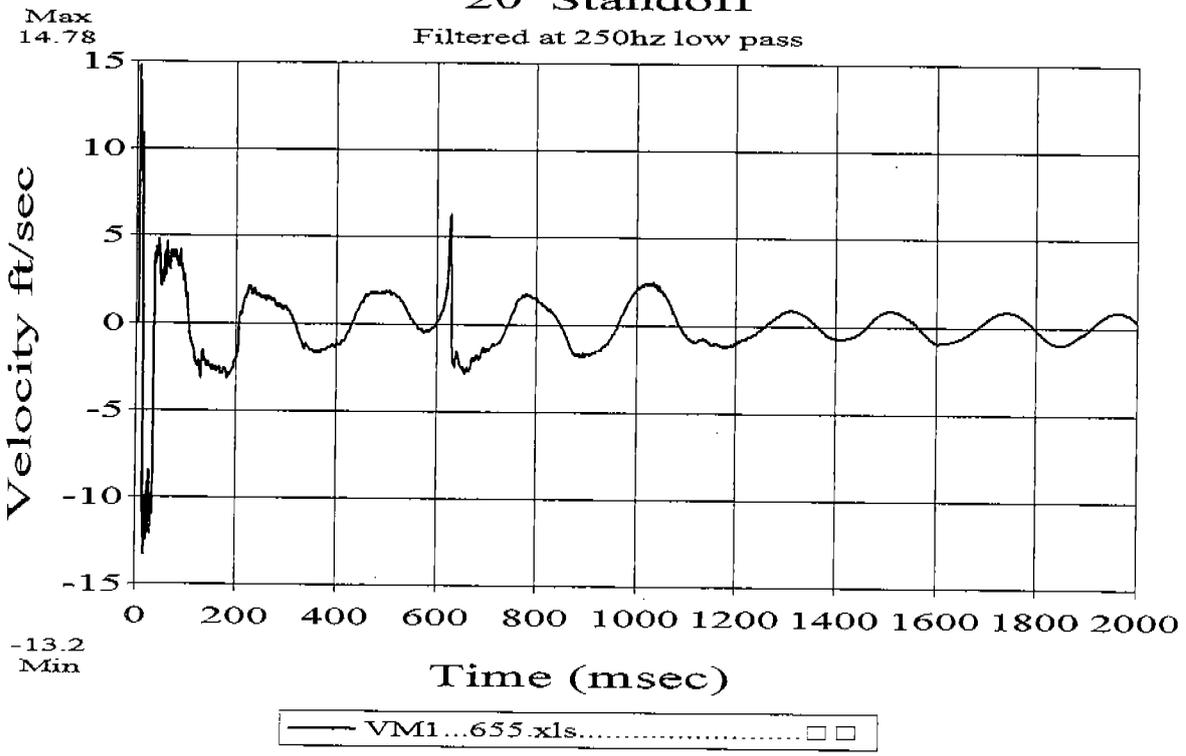
# Shot3\_Test4\_J655

20' Standoff



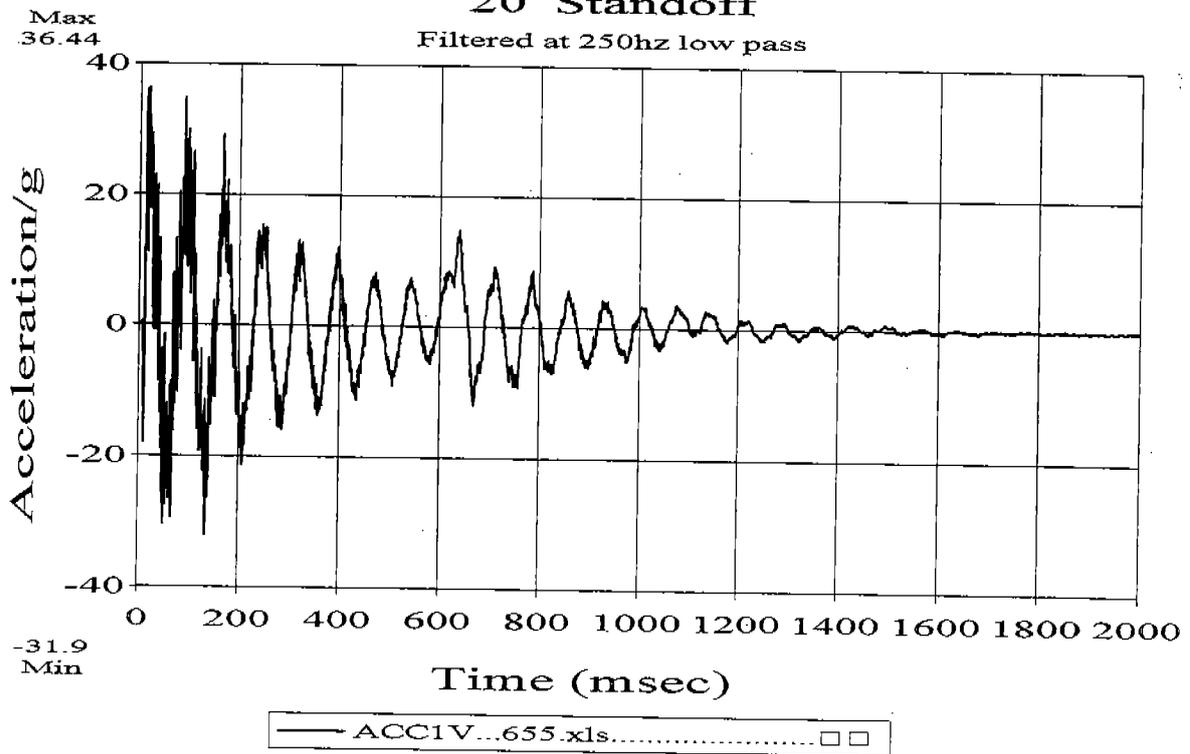
# Shot3\_Test4\_J655

20' Standoff



# Shot4\_Test4\_J655

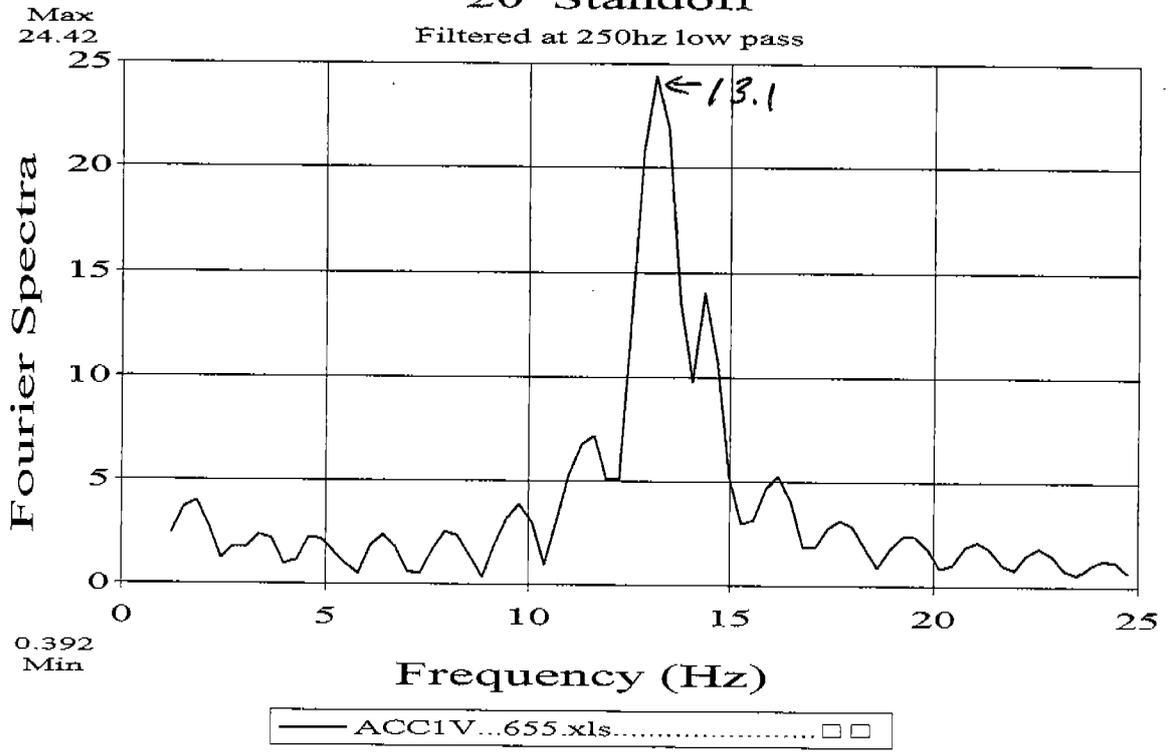
20' Standoff



# Shot4\_Test4\_J655

20' Standoff

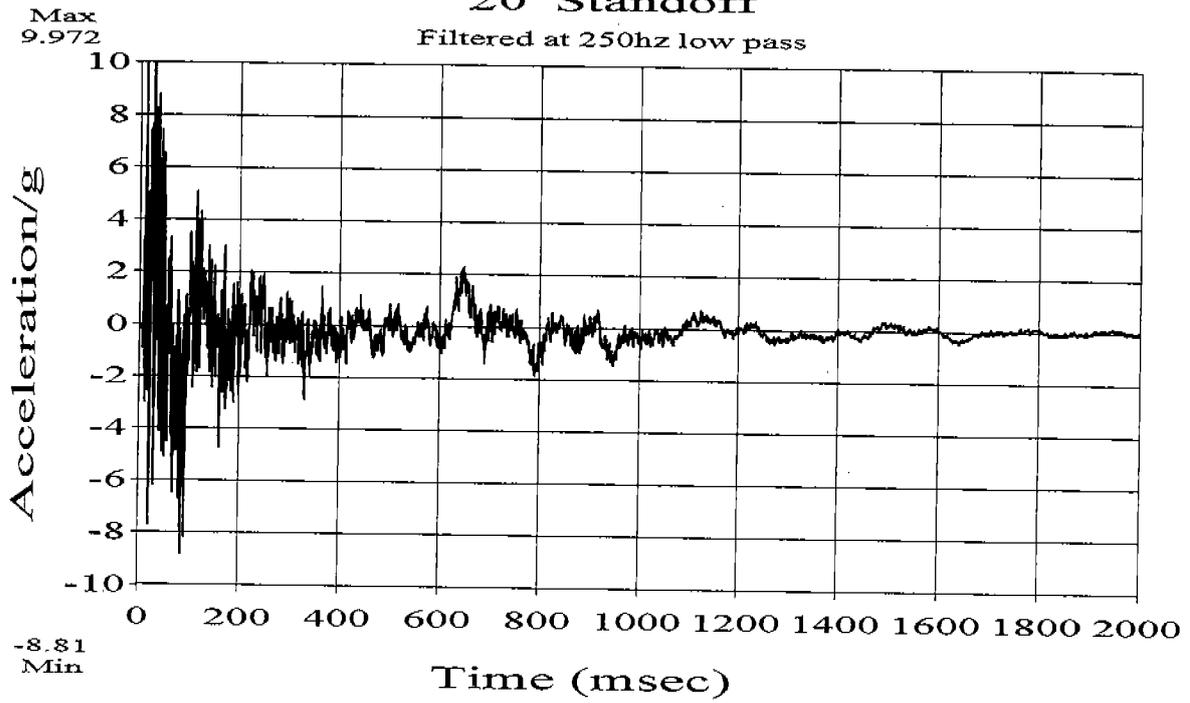
Filtered at 250hz low pass



# Shot4\_Test4\_J655

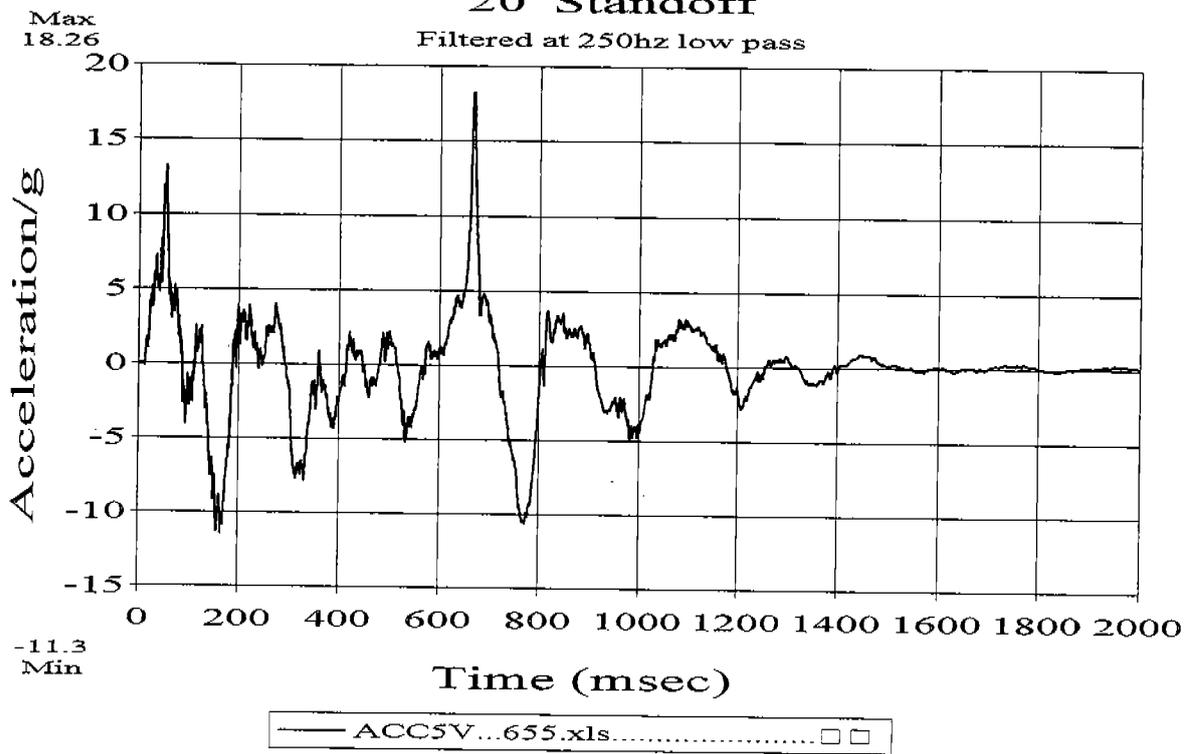
20' Standoff

Filtered at 250hz low pass



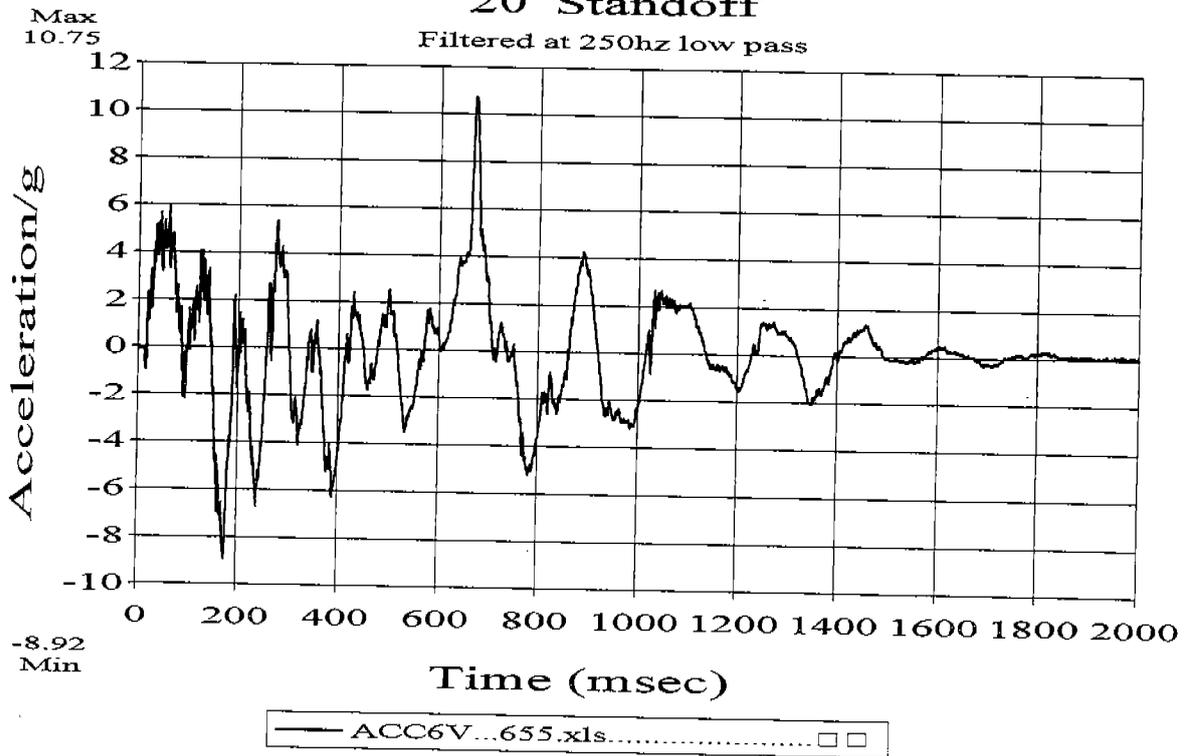
# Shot4\_Test4\_J655

20' Standoff



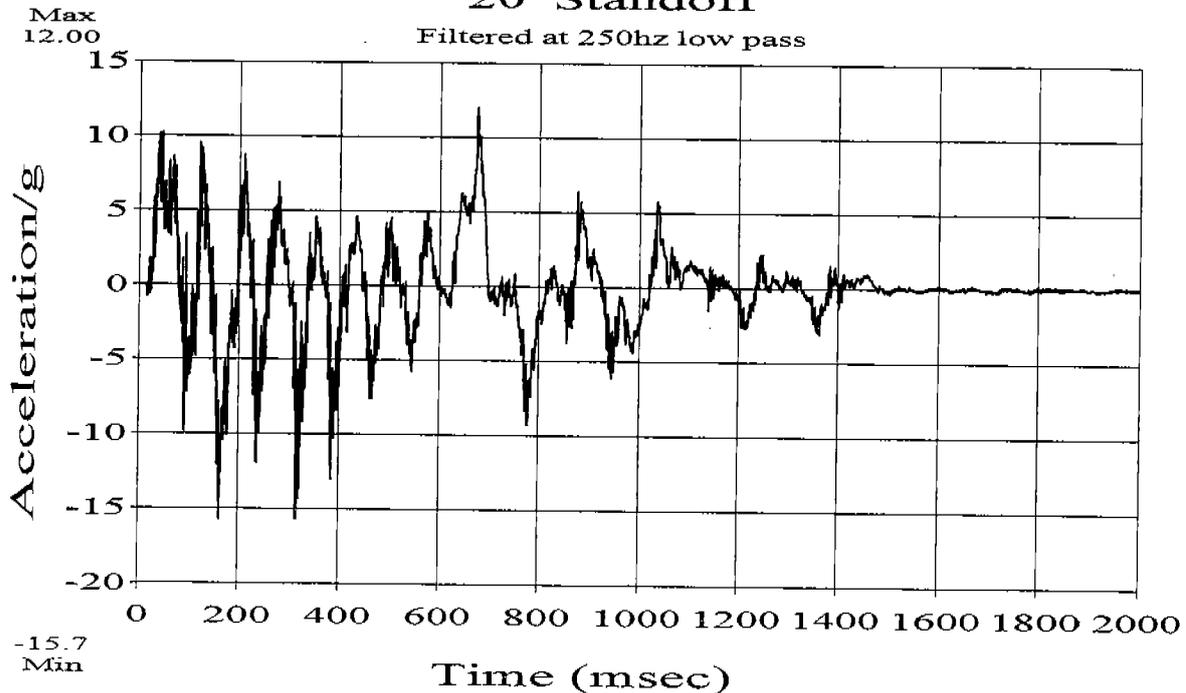
# Shot4\_Test4\_J655

20' Standoff



# Shot4\_Test4\_J655

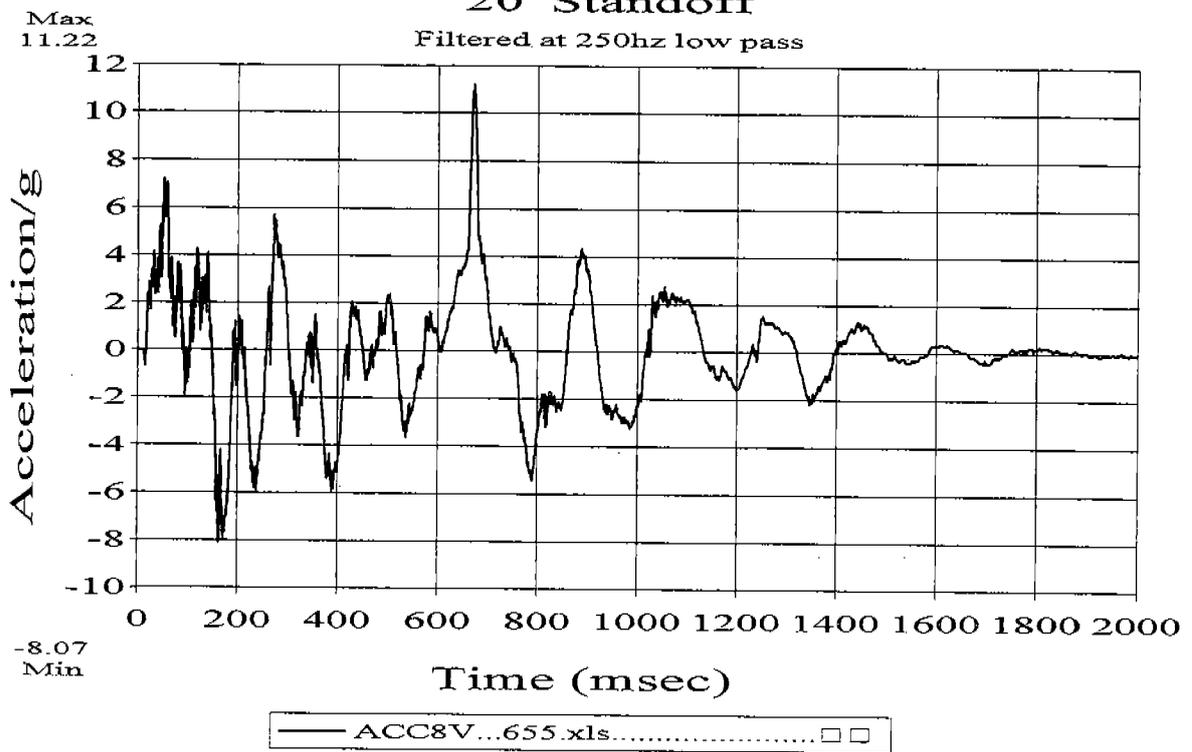
20' Standoff



— ACCTV...655.xls.....□□

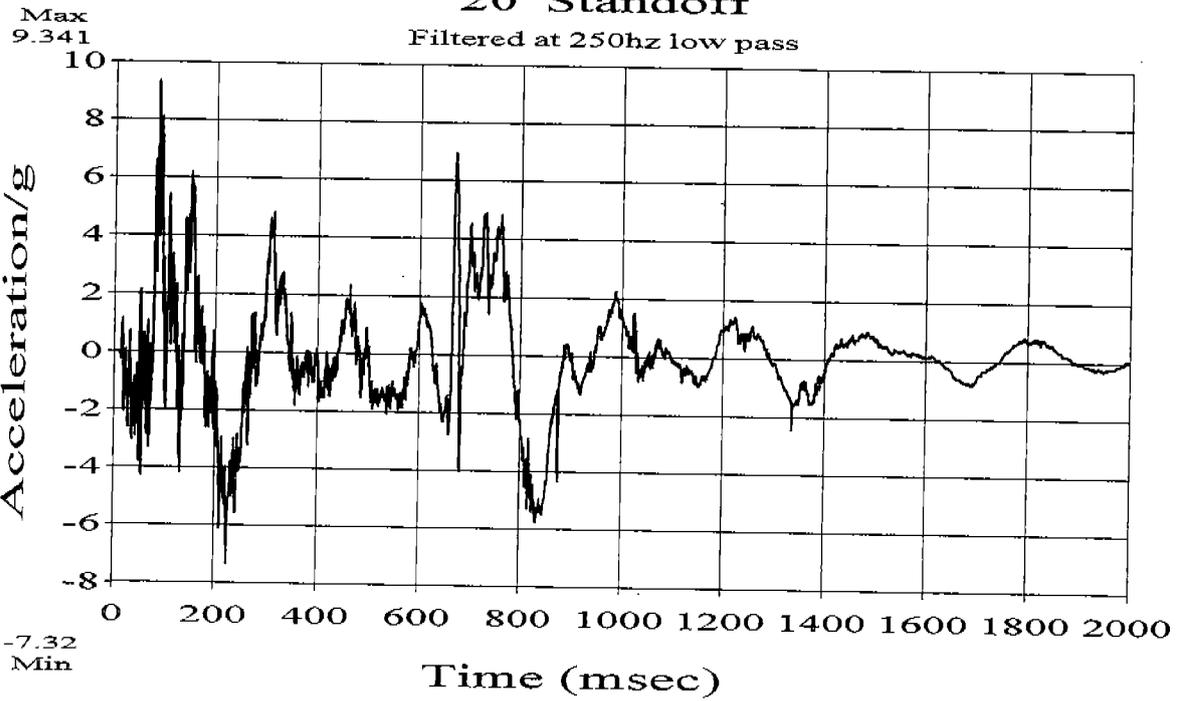
# Shot4\_Test4\_J655

20' Standoff



# Shot4\_Test4\_J655

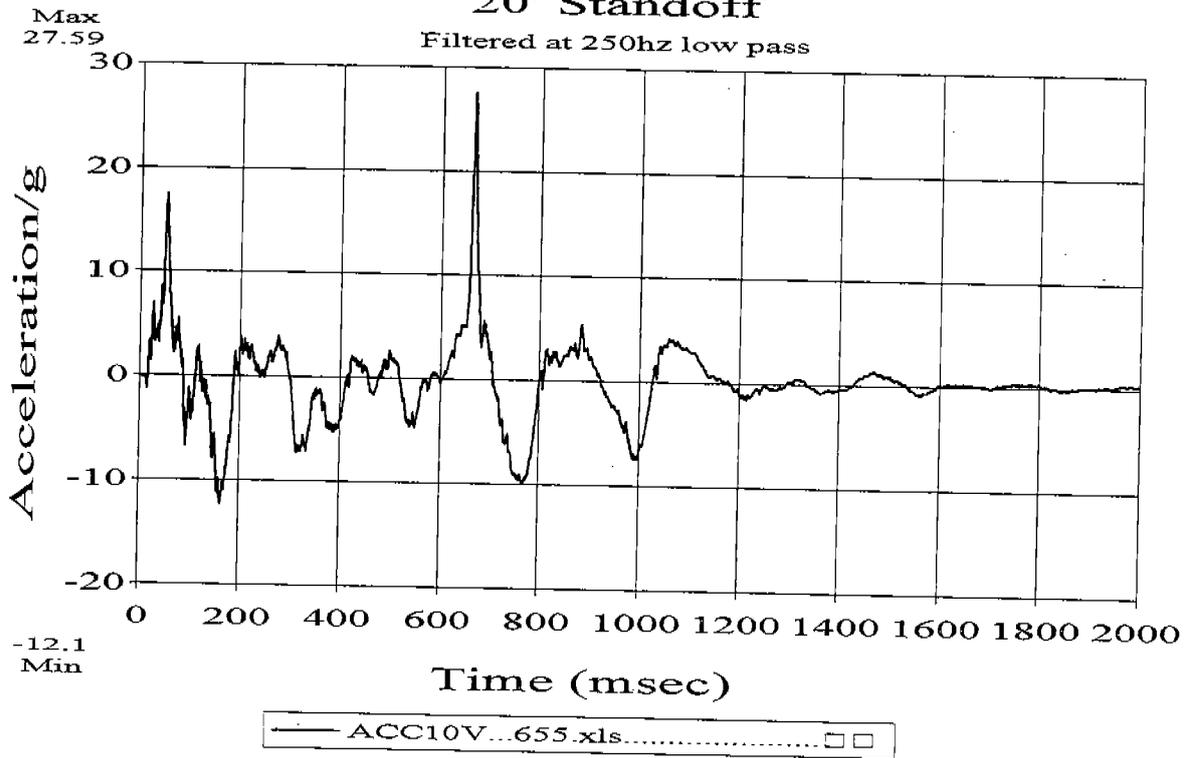
20' Standoff



— ACC9SS...655.xls..... □ □

# Shot4\_Test4\_J655

20' Standoff

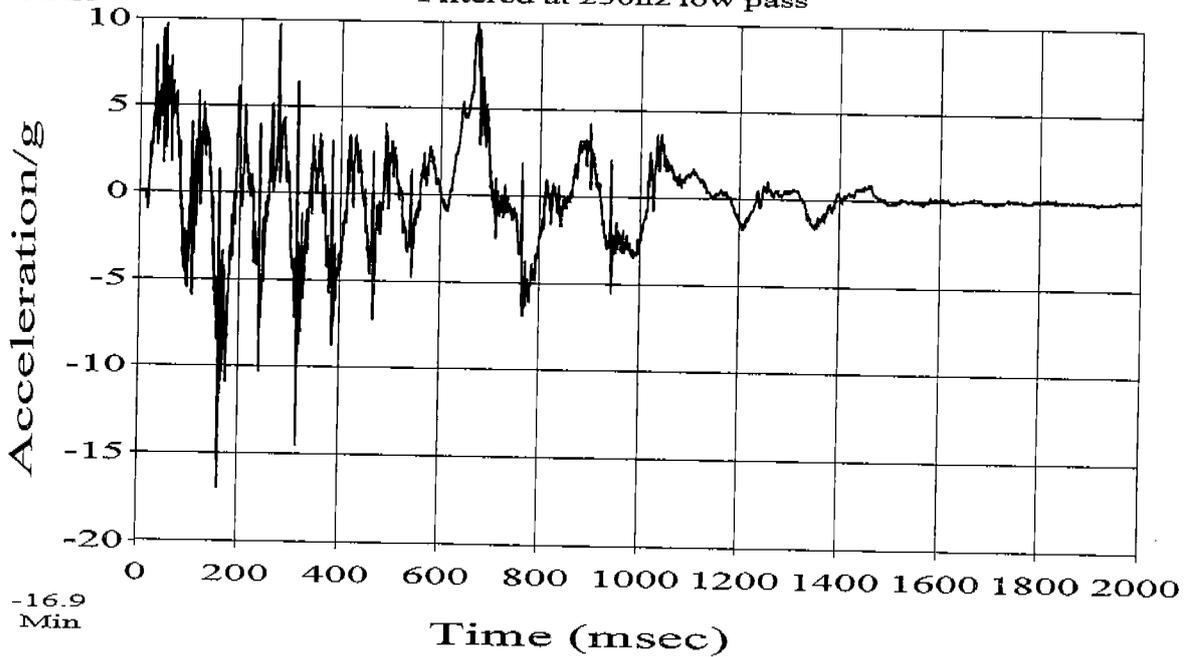


# Shot4\_Test4\_J655

20' Standoff

Max  
9.925

Filtered at 250hz low pass



-16.9  
Min

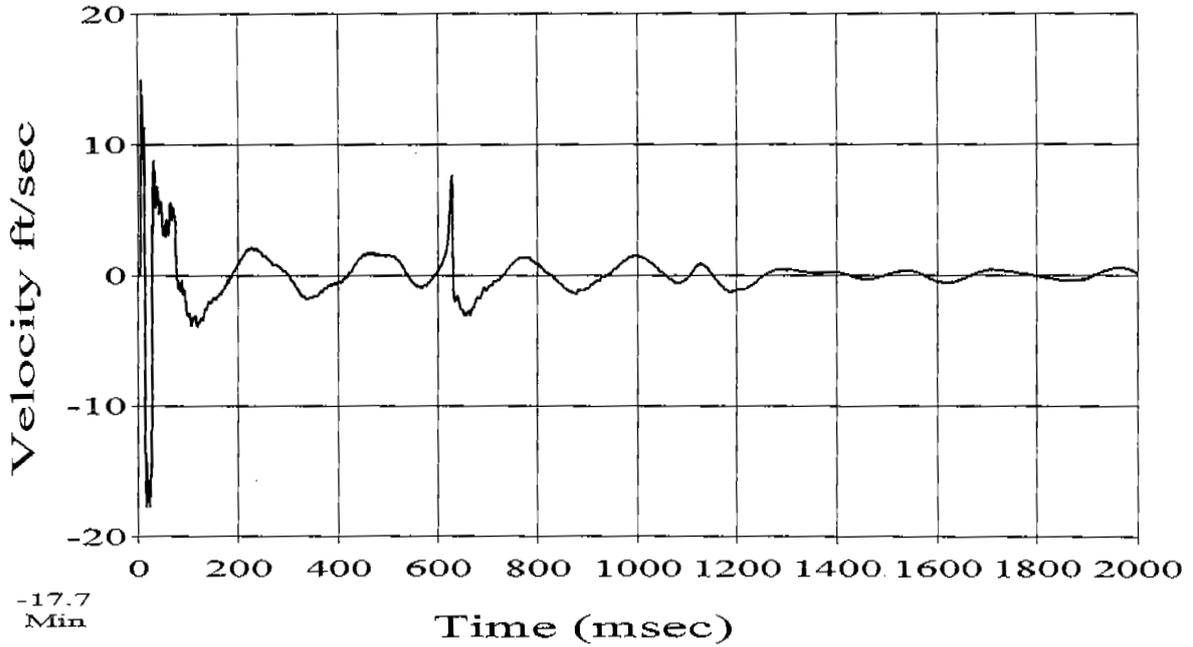
— ACC11V...655.xls..... □ □

# Shot4\_Test4\_J655

20' Standoff

Max  
14.93

Filtered at 250hz low pass



— VM1...655.xls..... □ □